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AIR FORCE FLIGHT DYNAMICS LABORATORY
DIRECTOR OF LABORATORIES
AIR FORCE SYSTEMS COMMAND
WRIGHT PATTERSON AIR FORCE BASE OHIO



A USER'S MANUAL FOR THE
SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS
COMPUTER PROGRAM

OCTOBER 1973

J. M. POTTER
R. A. NOBLE

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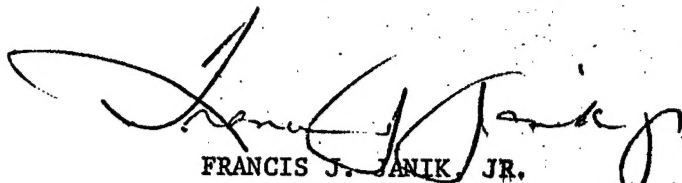
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FORWARD

This program was prepared by J. M. Potter of the Solid Mechanics Branch and R. A. Noble of the Experimental Branch, Structures Division, Air Force Flight Dynamics Laboratory. This work was conducted in-house under Project 1347, "Structural Testing of Flight Vehicles," Task 134704, "Structural Testing Criteria." This memo covers work accomplished over a time period of 1 October 1972 to 1 May 1973.

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This Technical Memorandum has been reviewed and is approved.



FRANCIS J. JANIK JR.
Chief, Solid Mechanics Branch
Structures Division

ABSTRACT

This report presents a detailed description of a computer program to calculate cumulative damage of notched structural members subjected to arbitrary spectra. The Sequence Accountable Fatigue Analysis computer program develops its sequence sensitivity by tracking residual stresses local to a notch throughout the spectrum of loads. Residual stress relaxation analysis is included to increase the generality of the results. An example spectrum and resulting cumulative damage analysis are illustrated.

TABLE OF CONTENTS

| SECTION | | PAGE |
|--------------|--|------|
| I | Introduction | 1 |
| II | Program Outline | 2 |
| III | Input Data Requirements | 6 |
| IV | Output Options | 9 |
| REFERENCES | | 10 |
| FIGURES | | 12 |
| APPENDIX I | Program List | 16 |
| APPENDIX II | Sample Problem with Input Data Listing | 37 |
| APPENDIX III | List of Computer Program Symbols and Definitions | 54 |
| APPENDIX IV | Fatigue Life Input Data for Several Materials | 60 |

SYMBOLS

| | |
|---------------------|---|
| σ_{res} | Residual stress |
| σ_{max} | Maximum local stress level |
| σ_{min} | Minimum local stress level |
| σ_{ys} | Yield stress |
| $\sigma_{res_{EQ}}$ | Equilibrium component of the residual stress |
| ϵ_t | Local strain, total |
| ϵ_e | Elastic component of total strain |
| ϵ_p | Plastic component of total strain |
| S_{max} | Maximum applied stress level |
| S_{min} | Minimum applied stress level |
| S_{mean} | Mean applied stress level, $(S_{max} + S_{min})/2$ |
| K_t | Elastic stress concentration Factor |
| D | Damage |
| I | Integer describing the level number |
| N_{EP} | Equilibrium period, number of cycles for the local stresses to return approximately to the equilibrium conditions following an overload |
| C_1 | Residual stress relaxation constant |
| E_1, E_2 | Relaxation function exponents |
| N | Number of cycles |
| E | Modulus of elasticity |
| N_f | Number of cycles of life at a given stress or strain cycling level |
| ϵ_p | Strain intercept at one reversal on a log ϵ_p -log life curve |
| m | Slope of the log ϵ_p -log life curve |

SECTION I

INTRODUCTION

Cumulative damage analyses based upon the local stress-strain behavior at a notch appear to be reasonably successful in anticipating trends in fatigue life behavior of notched specimens subjected to spectrum loading (1-6). The type of behavior that usually occurs is that peak tensile loads tend to increase the fatigue life and peak compressive loads tend to decrease the life of notched structures compared to structures experiencing load spectra not having those peaks (5,6). Local behavior analyses, such as those developed by Smith (7) and Neuber (8), help to explain this phenomenon as being a result of the tensile peak load creating a compressive residual stress at the notch and, conversely, the compressive peak creating a tensile residual stress. The change in life occurs because the residual stress state modifies the subsequent damage accumulation rates.

The Sequence Accountable Fatigue Analysis computer program was developed to incorporate the local stress-strain approach with a recent residual stress relaxation analysis (6) in order to improve the sequence sensitivity of cumulative damage analysis. This technical memorandum presents the details of the resultant computer program and an example of its use. The correlation of predictions made with this analysis to actual results of tests experiencing spectrum loading is presented by Potter (9) and Potter and Noble (10).

SECTION II

PROGRAM OUTLINE

The Sequence Accountable Fatigue Analysis traces the stress-strain behavior local to a notch throughout an applied load spectrum and calculates the damage based on the local experience. The computer program is divided generally into the four parts or modules outlined in Fig. 1.

The basic input data for the material, specimen geometry, fatigue behavior qualities and spectrum, are developed in Module I. Module I is discussed further in Section III. Module II takes the input information and determines the local stress-strain behavior. Module III references the Range Pair Counting Method Subroutine to cycle count the local stress spectrum. Module IV determines the damage in the local stress-strain spectrum.

The basic analyses used in Modules II, III and IV are presented below.

Module II - Local Stress-Strain Behavior - The analysis used during the determination of the local stress behavior during the spectrum of loading is a combination of analyses developed by Smith (7), Neuber (8) and Potter (5). Smith's simple analysis indicated that the residual stress could be approximated by assuming that the initial stress-strain behavior was elastic upon unloading following plastic flow. Thus, the residual stress could be calculated knowing the

maximum local stress and the maximum applied stress as in Eq. 1 and in Fig. 2.

$$\sigma_{res_i} = \sigma_{max_i} - K_t S_{max_i} \quad (1)$$

The Sequence Accountable Fatigue Analysis computer program currently incorporates elastic-perfectly plastic stress-strain behavior. Therefore, σ_{max_i} is equal to the yield stress. For the cycles immediately following the peak stress, the residual stress determined in Eq. 1 modifies the elastic solution as shown in Eqs. 2 and 3 (provided that the following maximum applied stress is less than S_{max} and that there is no change in the residual stress due to a minimum applied stress causing reversed yielding).

$$\sigma'_{max_i} = \sigma_{res_{i-1}} + K_t S_{max_i} \quad (2)$$

$$\sigma_{max_i} = \sigma_{res_{i-1}} + K_t S_{max_i} \quad (3)$$

The analysis developed by Neuber ⁽⁸⁾ has been extended to cyclic loading by Wetzell ⁽²⁾ and Wetzell, Morrow and Topper ⁽³⁾ and used by many others ^(1,4-6) primarily to determine local stress-strain behavior. It is used in this program only to calculate plastic strains occurring when the residual stress undergoes a step change. The plastic strain calculation routine is accessed only when the σ_{max_i} or σ_{min_i} terms in Eqs. 2 and 3 exceed tensile or compressive yield stress levels, respectively. Figure 3 illustrates the calculation of the plastic strain.

The local stress-strain behavior, according to Wetzell ⁽²⁾ is related to the applied load by Eq. 4

$$\sigma \cdot \epsilon = (K_t S_{t_{max}})^2 / E \quad (4)$$

The plastic strain can be found by subtracting the elastic component from the total strain.

$$\epsilon_p = \epsilon_t - \epsilon_e = (K_t S_{\max})^2 / E \cdot \sigma_{\max} - \sigma_{\max} / E$$

Therefore, the plastic strain associated with S_{\max_i} is given in Eq. 5.

$$\epsilon_{p_i} = (K_t S_{\max_i})^2 / E \sigma_{ys} - \sigma_{ys} / E \quad (5)$$

If a residual stress existed prior to this plastic strain excursion, the plastic strain associated with that prior excursion is subtracted from Eq. 5 as shown in Eq. 6.

$$\epsilon_{p_i} = (K_t S_{\max_i})^2 / E \sigma_{ys} - (\sigma_{ys} - \sigma_{res_{i-1}})^2 / E \sigma_{ys} \quad (6)$$

A similar calculation is made for plastic strains occurring during the minimum stress peak.

In the analysis developed by Potter ⁽⁶⁾ the residual stress cyclically relaxes toward zero or an equilibrium residual stress as shown in Fig. 4 according to Eq. 7.

$$\sigma_{res_{N=1,2,\dots}} = (\sigma_{res_{N=1}} - \sigma_{res_{EQ}}) \exp(N/N_{EP_i} \ln(0.1)) \quad (7)$$

The N_{EP} term, the Equilibrium Period, is dependent upon the applied stress and the Residual Stress Relaxation Constant.

$$N_{EP_i} = (C1 / (K_t S_{\max_i}^{E1} \cdot K_t S_{\max_i}^{E2})) \quad (8)$$

The Residual Stress Relaxation Constant, $C1$, has not yet been experimentally defined but should be a constant for a material.

Module III - Cycle Counting Method

After the local stress and plastic strain behavior is calculated, the local stress spectrum is Range Pair Counted using a computer program developed by Tischler. ⁽¹¹⁾

Module IV - Damage Calculation

Damage is calculated separately for the plastic strain excursions and the elastic stress spectrum. The damage is determined from the conventional $D = \sum \frac{n}{N}$ calculation. Damage from each of the plastic strain cycles is determined from the Coffin-Manson (12) form.

$$D_i = 1./N_{fi} = 1./(\epsilon_{pi}/\epsilon_f)^{1/c}$$

Damage from the elastic stress cycles is determined in a similar manner. The maximum and minimum local stress levels are sequentially compared to unnotched S-N data in a Modified Goodman Diagram format. Damage is summed, and failure of the coupon is defined as the event occurring when the summed damage equals unity.

SECTION III

INPUT DATA REQUIREMENTS

In general, each spectrum analyzed will require slightly different programming in order to get the load history into a usable format for the core program. The basic program requires a certain family of information before any analytical predictions can be made. Appendix I contains a program listing for the Sequence Accountable Fatigue Analysis. The subroutine CORE which accesses the subroutines having to do with RPCM, the Range Pair Counting Method, contains the basic analysis. Subroutine SAL reads the data input and then references subroutine CORE. The subroutine SAL shown is one in which a block of cycles is repeated with optional cycles. A list of the input data cards and the resulting analysis is given in Appendix II.

The specific data requirements are given below.

1. Stress-Strain Behavior - The stress-strain behavior is presumed to be elastic - perfectly plastic with the tensile yield stress being equal to the compressive yield stress. The yield stress value used is an average of the monotonic behavior generally being above the 0.2% yield value and below the engineering ultimate strength.

2. Residual Stress Relaxation - The residual stress relaxation behavior of Eq. 7 and 8 is characterized by C_1 , the Residual Stress Relaxation Constant and E_1 and E_2 , the relaxation equation exponents. The Residual Stress Relaxation Constant, C_1 , has not yet been adequately determined. It should be a material property if the relaxation function

is correct and must be assumed. A reasonably accurate estimate of the Residual Stress Relaxation Constant for aluminum material falls in the range of $5-20 \times 10^6$ (cycles) (Ksi)². Further experimentation on the part of the analyst should develop a C1 usable for his set of conditions until actual measurement of residual stress relaxation behavior defines the relaxation function and constants. At present E1 and E2 are considered to be equal to 1.0.

3. Specimen Geometry - The elastic K_t value (if available) is entered into the analysis. If that value is not available then an estimate from some other method may be used. In certain cases, a value may be determined from a constant amplitude fatigue test of a similar structure by fitting several values of K_t to the analysis and determining the best correlation as is done with the K_f solution. Once a stress concentration factor, K_t , is determined for a specimen, that value is not changed from test-to-test of the same coupon configuration.

4. Load Multiplier - Different spectra are presented for analysis in different manners. Some data are presented in percent of maximum stress, others in terms of nominal stress, and others in terms of bending moment. The value of the load multiplier defines the nominal stress history.

5. Cumulative Damage Analysis - The damage from the range-paired elastic stress spectrum is determined by calculating a simple $\frac{n}{N}$ value for each level and accumulating the total. The N_{f1} value is determined from unnotched coupon S-N data in the Modified Goodman Diagram format.

The program requires the input of four second order equations describing the maximum and minimum stress levels at lives of 10^4 , 10^5 , 10^6 and 10^7 cycles. The coefficients of the equations are derived by least square fitting the S-N data presented in the form of Eq. 9.

$$S_{\max} = A(I)S_{\min}^2 + B(I)S_{\min} + C(I) \quad (9)$$

The A, B, and C coefficients for several typical materials are presented in Appendix IV. The S-N data shown was derived from various sources but usually from the MIL-HDBK-5A ⁽¹³⁾. The C coefficients correspond to the maximum stress level at zero to maximum applied stress conditions on the unnotched coupons.

The damage from the plastic strain cycles is determined using the Coffin-Manson relation to calculate the N_{f1} value. The conventional plastic strain intercept at one reversal and the ϵ_p - life slope values are used in the analysis. Specific measured values from the literature are used when available and typical values when they are not available.

6. Analysis or Test Spectrum - The last information needed is the order and magnitude of application of the spectrum used in the test.

SECTION IV

OUTPUT OPTIONS

The Computer Program prints the following output in the process of the analysis.

1. Maximum and minimum applied stress and local stress response through the spectrum. Also printed out is the residual stress, equilibrium stress, applied cycles, and the equilibrium period.
2. The elastic local stress history as input into the Range Pair Subroutine and the resulting Range Paired spectrum.
3. The plastic strain occurrence during the spectrum and the damage associated with each strain reversal.
4. The accumulated damage associated with the plastic strains.
5. The Range Paired elastic stress spectrum and the damage associated with each level.
6. The accumulated damage associated with the current block of loading including the plastic strain damage and the total damage since the initiation of cycling.

At the option of the analyst, he can print out all the above items or only two. The IPRINT value controls what data is printed.

If IPRINT = 1, all six items are printed for each flight or block.

If IPRINT = 2, all items except 2. above are printed. /

If IPRINT = 3, only items 4. and 6. above are printed.

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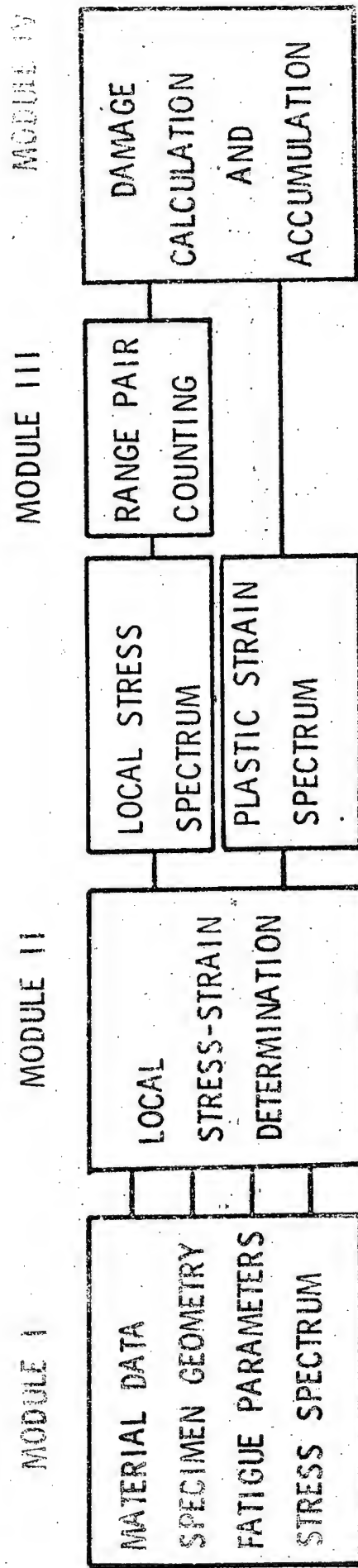


FIGURE 1. PROCEDURE USED IN THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

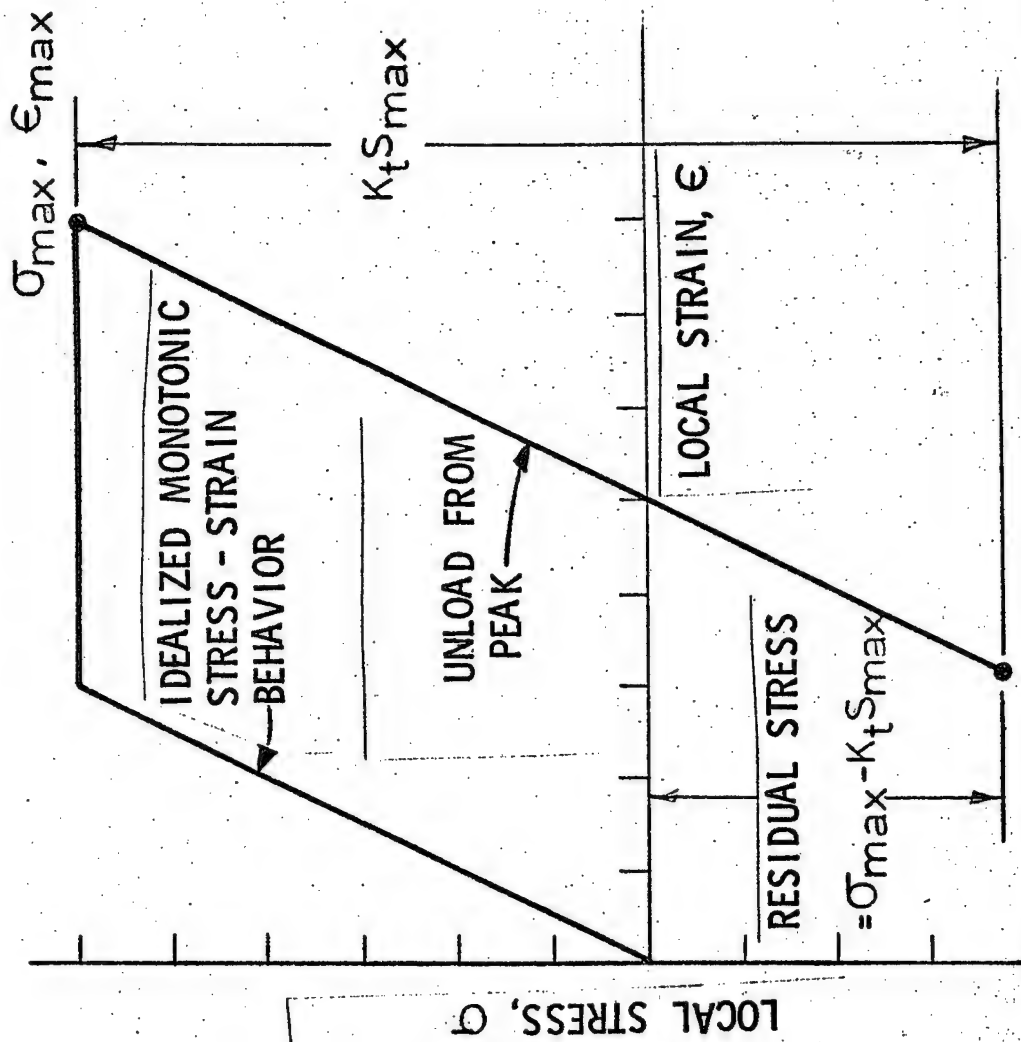


FIGURE 2. METHOD OF DETERMINING THE RESIDUAL STRESS FOLLOWING A PEAK LOAD

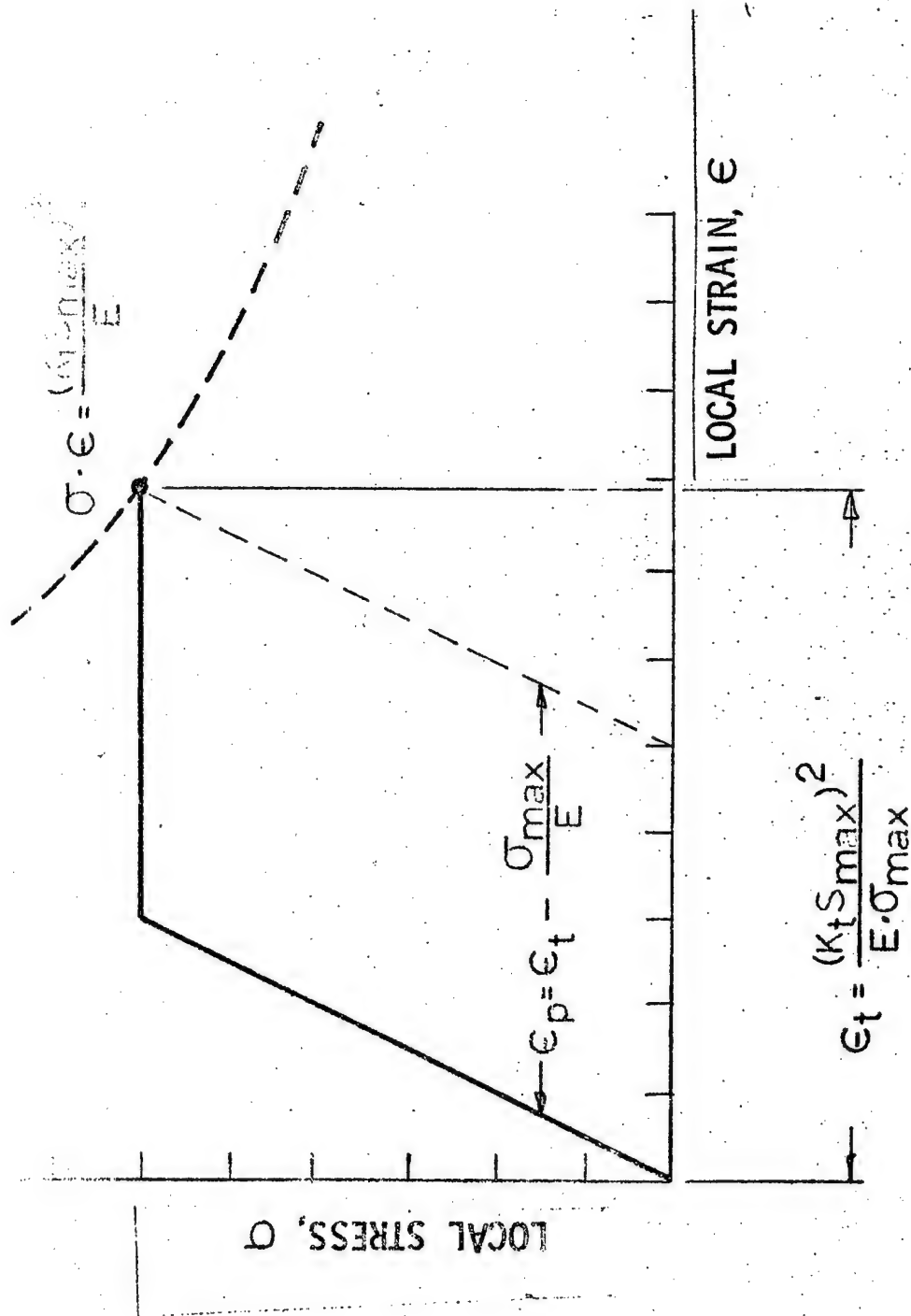
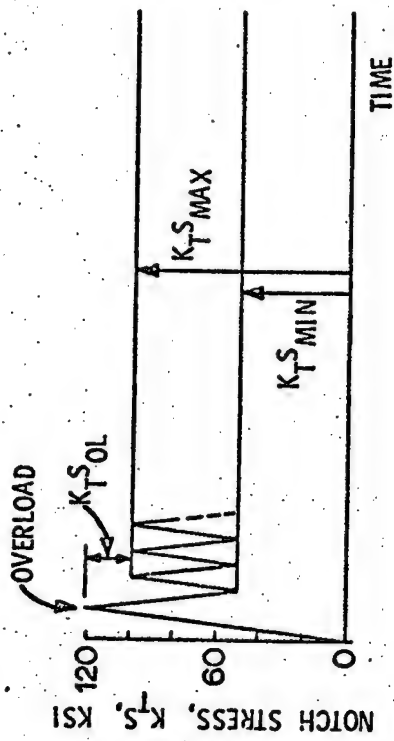
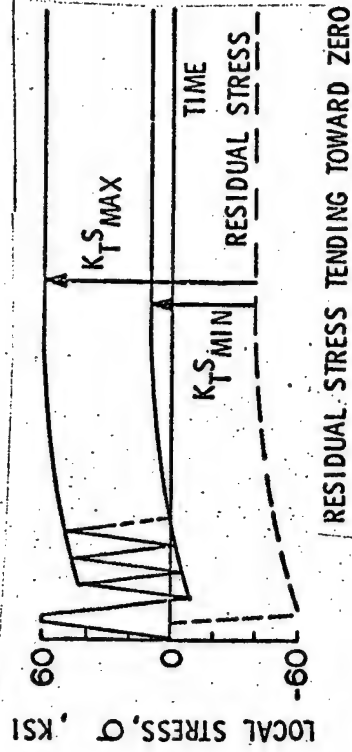


FIGURE 3. METHOD OF DETERMINING PLASTIC STRAIN LEVELS



APPLIED STRESS HISTORY



LOCAL STRESS RESPONSE

FIGURE 4. LOCAL STRESS RESPONSE FOR APPLIED CONSTANT AMPLITUDE LOADING WITH RESIDUAL STRESS RELAXATION

APPENDIX I PROGRAM LIST

PROGRAM SALV (INPUT, TAPE5=INPUT, OUTPUT, TAPE5=OUTPUT)

MODULE 1
INPUT ROUTINE FOR THE
SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

PROGRAM DEVELOPED BY JOHN M. POTTER AND ROBERT A. NOBLE
OF THE AIR FORCE FLIGHT DYNAMICS LABORATORY

MARCH 1973

ADDRESS: AFEDL/F33
WRIGHT-PATTERSON AFB OHIO 45433
TELEPHONE (513) 255-5104 OR 255-6106

INPUT

DATA CARD 1. NDECK = THE NUMBER OF DATA DECKS TO BE RUN SEQUENTIALLY.

IPRINT THE VALUE CONTROLLING THE WRITE STATEMENTS

1 PERMITS MAXIMUM PRINTOUT

2 SUPPRESSES RAINFLOW PRINTING

3 MAXIMUM SUPPRESSION OF PRINTOUT

IRPCM= THE VALUE CONTROLLING THE ENTRY INTO THE

RANGE PAIR COUNTING SUBROUTINE

1 ENTER RANGE PAIR COUNTING SUBROUTINE

2 SKIP RANGE PAIR COUNTING SUBROUTINE

FORMAT 314

EACH DATA DECK CONTAINS THE FOLLOWING CARDS --

CARD 1. TEST IDENTIFYING INFORMATION

FORMAT 8AB

CARD 2. TM = MATERIAL TYPE

IYS = TENSILE YIELD STRESS (KSI)

EPSD = LCF STRAIN INTERCEPT

COFMAN = INVERSE OF COFFIN-MANSON SLOPE

ELMOD = MODULUS OF ELASTICITY (KSI)

FORMAT 2AB,3F18.5,F10.2

CARDS 3....6. A(N) N=4,7 (A,B,C ARE COEFFICIENTS OF SECOND

ORDER LEAST SQUARE FIT OF S-N DATA,

(FOR CURVE OF 10**N CYCLES.)

B(N) (STMAX = A(N)*STMIN**2 + B(N)*STMIN + C(N))

C(N) (TITLE1, TITLE2 IDENTIFIES THE

TITLE1, TITLE2 IDENTIFIES THE

TITLE2 (SOURCE OF THE S-N DATA

N (PUNCHED IN COLUMN 72. FOR INFO.)

MATERIAL TYPE (COLUMNS 73-80. FOR INFO ONLY.)

FORMAT 3F18.5,2AB

CARD 7. C1 (CONSTANTS TO BE USED IN CALCULA-)

```

E1      .. (TION OF EQUILIARIUM PERIOD, ENEP.)
E2      (ENEP=CI/(KTSNAX**EI*KTSMEAN**E2))

```

```

IAKT = STRESS CONCENTRATION FACTOR USED THE FIRST
      TIME THROUGH THE PROGRAM
IZKI = STRESS CONCENTRATION FACTOR USED THE LAST
      TIME THROUGH THE PROGRAM
DELAKT = STEP CHANGE IN KT BETWEEN RUNS
FORMAT 3F10.5

```

UNBLOCK = NUMBER OF BLOCKS (NO. OF TIMES TO REPEAT LIST OF LOADS)

```

LEVEL = NUMBER OF LOADS
TYPE = NUMBER OF TYPES OF LOADS
FORMAT 3T10

```

TLL = LIMIT LOAD
FORMAT F18.5

```

NLEVEL + 10.
ISTEP(K) = K = THE KTH STEP (K=1,NLEVEL)
IISTYPE(K) = THE IDENTIFYING TYPE OF THE KTH LOAD
IISIN(K) = THE KTH MINIMUM DECIMAL FRACTION OF TLL)
IISMAX(K) = THE KTH MAXIMUM DECIMAL FRACTION OF TLL)
IENNK(K) = NUMBER OF CYCLES AT THE KTH LOAD
FORMAT 2(I4,2X),3(F48.5,1X)

```

```

+ 11,...,NLEVEL + 10 + NBLOCK.
(INBLOCK(JJ) = JJ = THE "JUTH" BLOCK (JJ=1,NBLOCK)
(NNN(JJ,1) = TYPE OF LOAD INCLUDED IN JUTH BLOCK
(NNN(JJ,2) = (THERE WILL BE ONE "NN" VALUE
(NNN(JJ, ) = (ON THE CARD FOR EACH DIFFERENT )
(FORMAT 915 (TYPE OF LOAD INCLUDED IN THE
(JUTH BLOCK

```

```
COMMON/MSAL/RNCYC(200),KPMAX,IPRINT
DIMENSION STIN(200),RTAX(200),RNN(200),ITYPE(200)
COMMON/MG031/NLEVL,IRPCA,IPL0T,ELMDO,IYS,EPSD,C0FYAN,C1,E2,E1,
RES(200),AXT,SUMCEN,SUMD
COMMON/MC032/STHIN(200),STMAX(200),ENN(200),A(10),B(10),C(10),
R(10),ISTEP(10),NN(200,10),IRLOCK(200)
```

```
WRITE(6,3) NOECK
FORMAT(1H1,I4,32H DATA DECKS ARE TO BE PROCESSED.)
IF(IRPCM.GE.2)GO TO 6
WRITE(6,11)
GO TO 13
```

WRITE(6,*)
FORMAT(25H NO COUNTING METHODS USED)
CONTINUE
DO 535 LMN = 1,NDECK
WRITE(6,5)
FORMAT(40H SEQUENCE ACCOUNTABLE FATI

```

115 10015,9)11,12,17,18,19,16,17,18
      FORMAT(8A3)
      WRITE(6,8)11,12,13,14,15,16,17,18
      FORMAT(16H1SPECTRUM FROM ,9A8)

```

INPUT OF DATA PECULIAR TO A MATERIAL

```

120 READ(5,10)TM1,TM2,TYS,EPSO,COFMAN,ELMON
      FORMAT(2A8,3F18.5,510.2)
      WRITE(6,12)TM1,TM2,TYS,EPSO,COFMAN,ELMON
      FORMAT(18H MATERIAL TYPE --,2A8,110H TENSILE YIELD STRESS (KSI)
      *--,F18.5,110H LCF STRAIN INTERCEPT =,F18.5,110H INVERSE OF COEFFI
      *N-MANSON SLOPE,F18.5,110H ELASTIC MODULUS =,F18.5)
      READ(5,14) (A(N),B(N),C(N),TITLE1,TITLE2,VE4,7)
      FORMAT(3F18.5,2A8)
      WRITE(6,16)

```

COEFFICIENTS OF SECOND ORDER LEAST SQUARE FIT OF S-N DA

```

130 *1A)
      WRITE(6,18)
      FORMAT(44H)
      WRITE(6,20)
      FORMAT(5X,5H LIFE,10X,5H A(1),14X,5H B(1),14X,5H C(1))
      WRITE(6,22) (A(N),B(N),C(N),N=4,7)
      FORMAT(3H 10**12,3F18.5)
      WRITE(6,24) TITLE1,TITLE2
      FORMAT(39HUNNOTCHED COUPON S-N DATA DERIVED FROM 728H INFORMATION
      *SUPPLIED FROM ,2AP)
      READ(5,14) C1,E1,E2
      WRITE(6,24)
      FORMAT(36HRESIDUAL STRESS RELAXATION FUNCTION)
      WRITE(6,26)
      FORMAT(74H)
      WRITE(6,28) C1,E1,E2
      FORMAT(13H*WHERE C1 =,E15.8,9H , E1 =,F10.3,10H AND E2 =,F10.
      *3)

```

88

INPUT OF DATA PECULIAR TO A SEQUENCE

```

150 READ(5,65)AKT
      FORMAT(F18.5)
      READ(5,32) NBLOCK,JLEVEL,NTYPE
      FORMAT(3I10)
      WRITE(6,34) NBLOCK,JLEVEL
      FORMAT(// 110,23H TIMES THROUGH BLOCK OF,110,6H LOADS)
      READ(5,35) TLL
      FORMAT(F18.5)
      WRITE(6,33) TLL
      FORMAT(// 5X,13H LOAD LIMIT =,F18.5)
      READ(5,36) (ISTEP(K),ITYPE(K),RTMIN(K),RTMAX(K),KEI,JLEVEL)
      FORMAT(14,2X,14,2X,F18.5,1X,F18.5,1X,F18.5,1X)
      WRITE(6,38)
      FORMAT(// 11H STEP TYPE,10X,6H STMIN,14X,6H STMAX,15X,4H ENN)

```

```
WRITE(6,36) (ISTEP(K), ITYPE(K), RTMIN(K), RTMAX(K), RNN(K), K=1, JLEVEL)
```

```
WRITE(6,39)
```

```
FORMAT(//47H NBLOCK TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE)
```

```
DO 42 JJ=1, NBLOCK
```

```
READ(5,40) IBLOCK(JJ), (NN(JJ, KK), KK=1, NTYPE)
```

```
WRITE(6,40) IBLOCK(JJ), (NN(JJ, KK), KK=1, NTYPE)
```

```
FORMAT(3I5)
```

```
CONTINUE
```

```
SUMENN=0.
```

```
SUMNC=0.
```

```
RES(1)=0.
```

```
WRITE(6,51) T1, T2, T3, T4, T5, T6, T7, T8
```

```
WRITE(6,51) (AKT)
```

```
FORMAT(77H AKT = , (F6.2))
```

```
WRITE(6,55) C1
```

```
FORMAT(//24H RELAXATION CONSTANT C1=, (F15.2))
```

```
IF(IRPCM, 5E-2) GO TO 59
```

```
WRITE(6,11)
```

```
FORMAT(36H0SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE)
```

```
*
```

```
CONTINUE
```

```
DO 1002 KFL=1, NBLOCK
```

```
JJJ=1
```

```
DO 60 J=1, JLEVEL
```

```
DO 70 KK=1, NTYPE
```

```
IF(NN(KFL, KK) .EQ. 0) GO TO 60
```

```
IF(ITYPE(J), EQ, NN(KFL, KK)) GO TO 150
```

```
CONTINUE
```

```
STMIN(JJJ)=RTMIN(J) * TLL
```

```
STMAX(JJJ)=RTMAX(J) * TLL
```

```
ENN(JJJ)=RNN(J)
```

```
JJJ=JJJ+1
```

```
CONTINUE
```

```
NLEVEL=JJJ-1
```

```
CALL CORE(KFL)
```

```
IPRINT=2
```

```
CONTINUE
```

```
597 CONTINUE
```

```
595 CONTINUE
```

```
596 CONTINUE
```

```
598 STOP
```

```
END
```

Figure 1. Schematic representation of the experimental design. The subjects were divided into two groups: the control group (CG) and the experimental group (EG). The CG was divided into two subgroups: the control group (CG) and the control group (CG). The EG was divided into two subgroups: the experimental group (EG) and the experimental group (EG). The subjects were divided into two groups: the control group (CG) and the experimental group (EG). The CG was divided into two subgroups: the control group (CG) and the control group (CG). The EG was divided into two subgroups: the experimental group (EG) and the experimental group (EG).

INFLUENCE OF SUBSTITUENT GROUPS

CORE PROGRAM OF THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

MODULE II

LOCAL STRESS AND STRAIN DETERMINATION

COMMON/40DEC01/SIGMAX(200),SIGMIN(200)
COMMON/MSAL/RNOCYC(200),RPMAX,IP2INT
DIMENSION PLSTRA(200),FM(210),AVSGMAX(200),AVSGMIN(200),EX(200),
ENRNCYC(200),ASX(200),ASHC(200),ASE(200)
COMMON/MC001/NLVEL,IRPCH,IPLOC,ELNOC,IYS,EPSC,CORFMAN,C1,E2,E1,
RES(200),AXT,SUMENN,SUMNC
COMMON/MC002/SIMTA(200),STHAX(200),FMIN(200),FMAX(200),C(10),C(10),
C(10),YST(200),NN(200,10),TLOCK(200)

1327

IF(UJ.GT.3)IPZINT=3.

Index

$$1277 = (6, 54)JJ$$

FORMAT(20H FLIGHT 02 BLOCK 17, 15)

IF (IPRINT,GE,3) GO TO 41

$$\text{WRT} = (5, 62)$$

... 156H S
... 156H S
... 156H S

CONFIDENTIAL

— ONTARIO

KPLOT=1

NO 570 J=1, N

$$t = j + 1$$
$$p(\text{std} \mid \text{std} \mid \text{std}) = 0.$$

DETROIT

--- PLASTIC ---
--- UELERLINE ---

0115474

$$A \text{--} \text{SMAX} = A < T * S$$
$$A_{SMIN} = A_{KT} * ST$$
$$A_{SX}(x, y, z) = (10.7 dx) x^5 y$$

ASW(XPLCT) ÷ 0

$$E = (107d) / 35V$$

KPLOT=KPLOT

$$ASHE \wedge N = (AS \cdot HE)$$
$$U = U$$

```
IF RES(I-T)  
IA=1
```

$$\overline{AA} = \overline{SS} = \overline{II}$$
$$PLST?A(J) =$$

U9=0

IF (RES(I-1))

4

1. The first step is to identify the problem or question that needs to be answered. This involves understanding the context and the specific requirements of the task.

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```
210 J3=-1
    AAA=ASMAX
    AAB=IYS-RES(I-1)
    IF (AAB.LE.0) GO TO 214
    60 AAB=0.
    212 PLSTRA(J)= AAA*AAA/(ELMOD*IYS)-AAB*AAB/(ELMOD*IYS)
    214 IF (JA+JB) 230,250,240
    220 RES(I)=IYS-ASMAX
    230 GO TO 230
    65 1240 RES(I)=IYS-ASHIN
    GO TO 230
    250 IF (JA) 260,260,270
    260 RES(I)=RES(I-1)
    GO TO 290
    70 RES(I)=-ASMEAN
    290 SIGNMAX(IRAIN)=RES(I)+ASMAX
    SIGNMIN(IRAIN)=RES(I)+ASMIN
    RNOYC(IRAIN)=ENN(J)
    IF (ASMAX.LE.IYS) GO TO 410
    75 ERES=-ASMAX+IYS
    GO TO 440
    410 IF (ASHIN.GE.-IYS) GO TO 430
    420 ERES=-ASHIN-IYS
    GO TO 440
    80 430 ERES=0.
    440 DIFF=RES(I)-ERES
    GO TO 446
    21
    85 CALCULATE RELAXATION FUNCTION
    ABMAX=ABS(ASMAX)
    ABMIN=ABS(ASMIN)
    ABMEAN=ABS(ASMEAN)
    IF (ABMAX.LT.1.) ABMAX=1.
    IF (ABMEAN.LT.1.) ABMEAN=0.5
    IF (ABMIN-ABMAX) 444,444,442
    90 442 ABM=ABMIN
    GO TO 446
    444 ABME=ABMAX
    446 ENEP=C1/(ABM**E1*ABMEAN**E2)
    IF (IPRINT.GE.3) GO TO 351
    WRITE(6,350) STMAX(J),STMIN(J),SIGMAX(IRAIN),SIGNMIN(IRAIN),
    *RES(I),ERES,ENN(J),ENEP,J
    350 FORMAT(6(F7.2,1X),F6.2,1X,E15.8,5X,I6)
    351 CONTINUE
    100
    105 CALCULATE RESIDUAL STRESS RELAXATION
    IRAIN=IRAIN+1
    ABOIF=ABS(OIF4)
    GO TO 350
    360 IF (ABOIF.LT.5.) GO TO 560
    370 IF (1000.*ENN(J).LT.ENEP) GO TO 560
    IF (ENN(J).LE.10.) GO TO 560
    110 380 NFLAG=0
```

005 00100 002

```

112      DELA2=10
113      IRAIN=IRAIN-1
114      DUMMY=EEN(J)
115      IF (DUMMY-ENEP) 470,460,460
116      DUMMY=DUMMY/2.
117      NFLAG=NFLAG+1
118      GO TO 450
119      CYCINT=DUMMY/10.
120      DO 530 K=1,10
121      DECK=FLOAT(K)
122      EN(K)=CYCINT*DECK
123      IF (K=10.1) GO TO 490
124      EX(K)=EXP(-2.303*EN(K-1)/ENEP)*EXP(-2.303*EN(K)/ENEP)
125      GO TO 500
126      EX(K)=1.+EXP(-2.303*EN(K)/ENEP)
127      CONTINUE
128      IF (NFLAG.EQ.0) GO TO 530
129      NFLAG=NFLAG+10
130      DO 520 K=11,NFLAG2
131      EN(K)=2.*DUMMY
132      EX(K)=EXP(-2.303*EN(K-1)/ENEP)*EXP(-2.303*EN(K)/ENEP)
133      DUMMY=2.*DUMMY
134      CONTINUE
135      DO 550 K=1,NFLAG2
136      AVSGMX(K)=ASHAX+ENEP*EX(K)/2.
137      AVSGMTR(K)=AVSGMX(K)-ASHAX+ASH11
138      SIGMINT(RAIN)=AVSGMTR(K)
139      SIGMAX(RAIN)=AVSGMX(K)
140      RNCYC(RAIN)=EN(K)
141      IF (K.EQ.1) GO TO 543
142      RNCYC(RAIN)=RNCYC(RAIN)-EN(K-1)
143      IF (IPRINT.GE.3) GO TO 561
144      WRITE (6,550) SIGMAX(RAIN),SIGMINT(RAIN),EN(K),RNCYC(RAIN)
145      FORMAT(16X,2(F7.2,1X),16X,F6.2,17X,F15.8)
146      CONTINUE
147      IRAIN=IRAIN+1
148      CONTINUE
149      RES(1)=QRES+QIF*EXP(-2.303*EEN(J)/ENEP)
150      CONTINUE
151      RES(1)=RES(J)
152      IN=IRAIN-1
153      *****
154      *****
155      *****
156      *****
157      *****
158      *****
159      *****
160      *****
161      *****
162      *****
163      *****
164      *****
165      *****
166      *****

```

MODULE III

CYCLE COUNTING TECHNIQUE

CALL SUBROUTINE TO RANGE PAIR COUNT SPECTRUM

IF (IRPCM.GT.1) GO TO 591

CALL RPSMINT

GO TO 592

```
591 CONTINUE
    KPMAX=IN
592 CONTINUE
170 C *****
171 C *****
172 C *****
173 C *****
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```

MODULE IV
DAMAGE ACCUMULATION CALCULATION

IF (IPRINT.GE.3) GO TO 552
WRITE(5,53)
FORMAT(776JH,LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGU
*E LIFE//10X,4HSTEP,10X,14HPLASTIC STRAIN,10X,10HMAX OR MIN,15X,6HD
*AMAGE)

552 CONTINUE
C CALCULATE DAMAGE FROM PLASTIC STRAIN CYCLES
SUMDEL=0.
DO 531 JKL=1,NLEVEL
AA=1.
IF (PLSTRA(JKL)) 532,531,533
532 AA=1.
533 PLSTRA(JKL)=AA*PLSTRA(JKL)
CYCLES=PLSTRA(JKL)/EPSD)**COFMAN
NAME1,CYCLES
SUMNC=SUMNC+DAM
SUMDEL=SUMDEL+DAM
IF (IPRINT.GE.3) GO TO 531
IF (AA) 535,535,537
535 WRITE(6,199)JKL,PLSTRA(JKL),DAM
199 FORMAT(10X,I4,12X,F10.5,15X,3HMIN,15X,E14.6)
GO TO 531
537 WRITE(6,219)JKL,PLSTRA(JKL),DAM
219 FORMAT(10X,I4,12X,F10.5,15X,3HMAX,15X,E14.6)
531 CONTINUE
WRITE(5,541) SUMDEL
541 FORMAT(59X,29H DAMAGE FROM PLASTIC STRAINS=,E15.8)
IF (IPRINT.GE.3) GO TO 536
WRITE(6,13)
13 FORMAT(/16X,15H SIGNMAX SIGNMIN,18X,6H RMCYC,20X,25H CYCLES
* ENH/CYC)
536 CONTINUE
C CALCULATE ELASTIC CYCLE DAMAGE FROM LEAST SQUARE FITTED S-N DATA
(MODIFIED GOODMAN DIAGRAM FORMAT)
DO 600 JKL=1,KPMAX
IYYS=IYYS/5.
X=SIGNMIN(JKL)
Y=SIGNMAX(JKL)
IF (Y-X.LT.1.6*YYS) GO TO 310
CYCLES=10.**4.


```

225      60 TO 340
226      IF (W,GE,TVST) 60 TO 370
227      CYCLES=10.**9.
228      60 TO 340
229      N=4
230      DO 330 M=1,4
231      R(N)=AIN)**2+3(N)*X+C(N)-Y
232      N=N+1
233      CONTINUE
234      IF (R(7)*R(4)) 335,338,334
235      A9P7=185(R(7))
236      A9P4=105(R(4))
237      IF (A9P2,15,AB34) 63 TO 335
238      EXPD=4+R(4)/(R(4)-R(5))
239      60 TO 339
240      EXPD=7+R(7)/(R(6)-R(7))
241      60 TO 339
242      SUM2=R(4)+R(5)+R(6)+R(7)**2
243      SUM22=R(4)**2+R(5)**2+R(6)**2+R(7)**2
244      SUM23=R(4)**3+R(5)**3+R(6)**3+R(7)**3
245      SUM24=R(4)**4+R(5)**4+R(6)**4+R(7)**4
246      SUM2N=4.*R(4)+5.*R(5)+6.*R(6)+7.*R(7)
247      SUM2N4.*R(4)**2+5.*R(5)**2+6.*R(6)**2+7.*R(7)**2
248      DEL1=4.*SUM2*SUM2-SUM24.*SUM2**2
249      DEL2=SUM2*SUM22*SUM23-SUM24*SUM2**3
250      DEL3=SUM2*SUM22*SUM23-SUM24*SUM2**3
251      D01=22.*SUM2*SUM24-22.*SUM21**2
252      D02=SUM22*SUM23*SUM2N-SUM2N*SUM24*SUM2H
253      D03=SUM2*SUM23*SUM22H-SUM22H*SUM22**2
254      EXPD=1001+D01+D03/(DEL1+DEL2+DEL3)
255      CYCLES=10.**EXPD
256      IF (EXPD,LE,4.7) CYCLES=10.**4.
257      ENNCYC(JKL)=ENNCYC(JKL)/CYCLES
258      SUMNC=SUMNC+ENNCYC(JKL)
259      SUMDEL=SUMDEL+ENNCYC(JKL)
260      IF (IP2INT,GE,3) 50 TO 600
261      WRITE(6,599) Y,X,ENNCYC(JKL),CYCLES,ENNCYC(JKL)
262      FORMAT(16X,2(F7.2,1X),16X,F6.0,17X,2(E15.8,1X))
263      CONTINUE
264      WRITE(6,593)SUMDEL
265      FORMAT(/69X,21H DAMAGE PER THIS SET=,E15.3)
266      WRITE(6,575) SUMNC
267      FORMAT(/69X,18H TOTAL ENN/CYC =,E15.8)
268      END

```


311 CONTINUE

NPKSN = NPKSN - JMAS

C RANGE PAIR COUNT THE ADJUSTED LOAD SPECTRUM

115

C

6000 I = 1

KB = 0

L = JMAX

KMIN = 0

KMAX = 0

LR = 0

K31 = 0

1 IF (RNCYC(I) .GT. 1.0) GO TO 400

IF (KB .NE. 0) GO TO 5

X1 = SIGMAX(I)

X2 = SIGMIN(I)

IND1 = NSTEP(I)

IND2 = IND1

I = I + 1

KB = 1

GO TO 1

5

X3 = SIGMAX(I)

X4 = SIGMIN(I)

IND3 = NSTEP(I)

IND4 = IND3

KMIN = 1

KMAX = 0

K31 = 0

IF (RNCYC(I) .EQ. 1.0) GO TO 6

KEY = 1

KIND = 1

GO TO 415

6

KEY = 0

CYCNO = AINT(RNCYC(I)*0.5)

CALL DECIDE(X1,X2,X3,X4,KEY,I,CYCNO,KCYGEN)

1000 GO TO (10,10,30),KCYGEN

10

K9 = 1

I = I + 1

IF (KMIN .NE. 1) GO TO 36

IF (I .LE. NPKSN) GO TO 5

RES(LR+1) = X1

RES(LR+2) = X2

INDEX(LR+1) = IND1

INDEX(LR+2) = IND2

LRMAX = LR + 2

GO TO 2000

30

IF (KMIN .NE. 1) GO TO 35

I = I + 1

IF (I .LE. NPKSN) GO TO 31

RES(LR+1) = X1

RES(LR+2) = X2

RES(LR+3) = X3

INDEX(LR+1) = IND1

INDEX(LR+2) = IND2

INDEX(LR+3) = IND3

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```
2000 LMAX = 1
      IF (LRMAX .LT. 4) GO TO 5000
      IF (NCYNO .EQ. 0) GO TO 5000

225      C RANGE PAIR COUNT OF RESIDUE SPECTRUMS
      C
      NRES = NRES + 1
      CALL DECRES(LRMAX, NCYNO)
      GO TO 2000

230      C 5000 IF (LRMAX .LE. 1) GO TO 3000
      C
      C COUNT THE LAST RESIDUE SPECTRUM - RANGE PAIR COUNTING WILL YIELD N
      C ADDITIONAL CYCLES
      C
      KK = 0
      RESMAX = RES(1)
      RESMIN = RES(1)
      IMAX = 1
      IMIN = 1
      DO 500 I = 2, LRMAX
      IF (RES(I) .LT. RESMAX) GO TO 490
      RESMAX = RES(I)
      IMAX = I
      GO TO 500
490 IF (RES(I) .GT. RESMIN) GO TO 500
      RESMIN = RES(I)
      IMIN = I
      CONTINUE
      CALL CYCRES(RESMAX, RESMIN, 1.0, INDEX(IMAX))
      KK = KK + 1
      J = IMAX - 2
      IF (J .LE. 0) GO TO 550
      CALL CYCRES(RES(J), RES(J+1), 1.0, INDEX(J))
      KK = KK + 1
      IMAX = J
      GO TO 510
510 J = IMAX + 2
      IF (J .GT. LRMAX) GO TO 575
      CALL CYCRES(RES(J-1), RES(J), 1.0, INDEX(J-1))
      KK = KK + 1
      IMIN = J
      GO TO 550
575      KMAX = KK
      LMAX = L
      C
      C SORT THE ANALYSIS SPECTRUM TO PRODUCE THE RANGE PAIR COUNTED SPECT
      C
      3000 KP = 0
      DO 605 JJ = 1, MPKS
      KC = 0
      DO 600 I = 1, LMAX
      IF (NNSTEP(I) .NE. JJ) GO TO 600
      KP = KP + 1
      KC = KC + 1
      NSTEP(KP) = KP
      605
      600
      600
```


SUBROUTINE DECIDE

```
1151 IF (LIND .EQ. 1) GO TO 1153
IF (IND3 .NE. IND4) GO TO 1153
RNECYC(L) = RNECYC(L) + CYCNO - 2.0
CYCNO = 1.0
60 1152 IF (KMAX .NE. 1) GO TO 111
X3 = SIGMIN(I)
IND3 = NSTEP(I)
IF (CYCNO .NE. 0.0) GO TO 112
KMIN = 1
KMAX = 0
KCYGEN = 3
RETURN
1200 IF (CYCNO .EQ. 0.0) RETURN
CYCNO = CYCNO - 1.0
X3 = SIGMAX(I)
X4 = SIGMIN(I)
KFIRST = 1
GO TO 113
111 X3 = SIGMAX(I)
X4 = SIGMIN(I)
IF (KFIRST .NE. 0) GO TO 113
CYCNO = CYCNO - 1.0
KFIRST = 1
113 IND3 = NSTEP(I)
IND4 = IND3
KMIN = 1
KMAX = 0
GO TO 11
1500 IF (KMAX .NE. 0) GO TO 1510
IF (CYCNO .EQ. 0.0) RETURN
CYCNO = CYCNO - 1.0
112 X4 = SIGMAX(I)
IND4 = NSTEP(I)
KMAX = 1
KMIN = 0
GO TO 11
1510 X4 = SIGMIN(I)
IND4 = NSTEP(I)
KMAX = 0
KMIN = 1
GO TO 10
END
```


SUBROUTINE DECRES(LRMAX,NGYND)
COMMON/MODER/RES(1400),INDEX(1400),IND1,IND2,IND3,IND4,KIND

5 THIS SUBROUTINE DECIDES WHETHER OR NOT THE ELEMENTS OF THE RESIDUE
SPECTRUM SATISFY THE RANGE PAIR COUNTING CONDITIONS

K = J
NGYND = 0

X1 = RES(1)

X2 = RES(2)

X3 = RES(3)

X4 = RES(4)

IND1 = INDEX(1)

IND2 = INDEX(2)

IND3 = INDEX(3)

IND4 = INDEX(4)

J = 6

10 IF (X2.GT. X1) GO TO 100

IF (X2.LT. X4.OR. X3.GT. X1) GO TO 500

150 IF (X2.GT. X3) GO TO 151

CALL CYCRES(X3,X2,1.0,IND3)

GO TO 152

151 CALL CYCRES(X2,X3,1.0,IND2)

152 NGYND = NGYND + 1

X1 = X1

X2 = X4

IND2 = IND4

IF (J.EQ. LRMAX) GO TO 300

IF ((J+1).EQ. LRMAX) GO TO 315

X3 = RES(J+1)

X4 = RES(J+2)

IND3 = INDEX(J+1)

IND4 = INDEX(J+2)

J = J+2

GO TO 10

100 IF (X2.GT. X4.OR. X3.LT. X1) GO TO 500

GO TO 150

500 K = K + 1

RES(K) = X1

INDEX(K) = IND1

J = J + 1

IF (J.GT. LRMAX) GO TO 330

X1 = X2

X2 = X3

X3 = X4

X4 = RES(J)

IND1 = IND2

IND2 = IND3

IND3 = IND4

IND4 = INDEX(J)

GO TO 10

300 K = K + 1

RES(K) = X1

RES(K+1) = X2

CORE 679

CORE 679

CORE 680

CORE 681

CORE 682

CORE 683

CORE 684

CORE 685

CORE 686

CORE 687

CORE 688

CORE 689

CORE 690

CORE 691

CORE 692

CORE 693

CORE 694

CORE 695

CORE 696

CORE 697

CORE 698

CORE 699

CORE 700

CORE 701

CORE 702

CORE 703

CORE 704

CORE 705

CORE 706

CORE 707

CORE 708

CORE 709

CORE 710

CORE 711

CORE 712

CORE 713

CORE 714

CORE 715

CORE 716

CORE 717

CORE 718

CORE 719

CORE 720

CORE 721

CORE 722

CORE 723

CORE 724

CORE 725

CORE 726

CORE 727

CORE 728

CORE 729

CORE 730

CORE 731

CORE 732

SUBROUTINE DECRES

CDC 6600 FTN V3.0-351A OPT=1

09/15/73

09.20.05.

PAGE 2

INDEX(K) = IND1

INDEX(K+1) = IND2

LRMAX = K + 1

RETURN

CORE 733

CORE 734

CORE 735

CORE 736

CORE 737

CORE 738

CORE 739

60

315 K = K + 1

RES(K) = X1

RES(K+1) = X2

RES(K+2) = RES(J+1)

INDEX(K) = IND1

INDEX(K+1) = IND2

INDEX(K+2) = INDEX(J+1)

LRMAX = K + 2

RETURN

CORE 740

CORE 741

CORE 742

CORE 743

CORE 744

CORE 745

70

330 K = K + 1

RES(K) = X2

RES(K+1) = X3

RES(K+2) = X4

INDEX(K) = IND2

INDEX(K+1) = IND3

INDEX(K+2) = IND4

LRMAX = K + 2

RETURN

END

CORE 746

CORE 747

CORE 748

CORE 749

CORE 750

CORE 751

CORE 752

CORE 753

CORE 754

CORE 755

75

35

```

SUBROUTINE CYCLES(Y1,Y2, CYCPF,NSTPPP)
COMMON/4000/CYCLES(900,2),RNECYC(900),NNSTEP(900)
COMMON/4000/L,LIND
5  C THIS SUBROUTINE GENERATES CYCLES FOR THE ANALYSIS SPECTRUM FROM DA CORE
   C SUPPLIED BY SUBROUTINE DEGRS
   C
   C L = L + 1.
   C CYCLE(L,1) = Y1
   C CYCLE(L,2) = Y2
10  C RNECYC(L) = CYCPF
   C NNSTEP(L) = NSTPPP
   C RETURN
   C END

```

APPENDIX II

SAMPLE PROBLEM WITH INPUT DATA LISTING

1. DATA DECKS ARE TO BE PROCESSED.

SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE
SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

SPECTRUM FROM B-1 SPECTRUM -- TRUNCATION LEVEL 270.000 CYCLES
 MATERIAL TYPE -- 2210-1051 AL

TENSILE YIELD STRESS (KSI) -- 55.0000

LOG STRAIN INTERCEPT = .40000

INVERSE OF COFFIN-MANSACK SLOPE -1.83500

ELASTIC MODULUS = 10000.00000

COEFFICIENTS OF SECOND ORDER LEAST SQUARE FIT OF S-N DATA

SMAX = A(1)*SMIN**2 + B(1)*SMIN + C(1)

| LIFE | A(1) | B(1) | C(1) |
|-------|---------|--------|----------|
| 10**4 | -.00217 | .22041 | 55.82462 |
| 10**5 | -.00178 | .33217 | 48.26657 |
| 10**6 | -.00149 | .46283 | 39.66455 |
| 10**7 | -.00243 | .64199 | 31.70955 |

UNNOTCHED COUPON S-N DATA DERIVED FROM
 INFORMATION SUPPLIED FROM NORTH AMERICAN

RESIDUAL STRESS RELAXATION FUNCTION

ENEP = C1/(KTSMAX**E1 + KTSMEAN**E2)

WHERE C1 = .25000000E+07, E1 = 1.0000 AND E2 = 1.0000

7 TIMES THROUGH BLOCK OF 44 LOADS

LOAD LIMIT = 26.90000

| STEP | TYPE | SMIN | STHAX | ENN |
|------|------|---------|---------|----------|
| 1 | 1 | -.14800 | -.02900 | 1.00000 |
| 2 | 3 | .58400 | 1.00000 | 1.00000 |
| 3 | 2 | .58400 | .89500 | 1.00000 |
| 4 | 1 | .58400 | .68400 | 1.00000 |
| 5 | 1 | .45500 | .58400 | 1.00000 |
| 6 | 1 | .64800 | .59300 | 1.00000 |
| 7 | 1 | .39500 | .69900 | 1.00000 |
| 8 | 1 | .48800 | .56300 | 1.00000 |
| 9 | 1 | .23500 | .59900 | 1.00000 |
| 10 | 1 | .50900 | .62700 | 2.00000 |
| 11 | 1 | .43400 | .59900 | 1.00000 |
| 12 | 1 | .36400 | .69000 | 1.00000 |
| 13 | 1 | .13300 | .36400 | 1.00000 |
| 14 | 1 | .36400 | .48200 | 6.00000 |
| 15 | 1 | .30400 | .36400 | 1.00000 |
| 16 | 1 | .30700 | .75600 | 1.00000 |
| 17 | 1 | .31600 | .52100 | 1.00000 |
| 18 | 2 | .54500 | .70800 | 1.00000 |
| 19 | 1 | .33400 | .61700 | 1.00000 |
| 20 | 1 | .45100 | .51200 | 7.00000 |
| 21 | 1 | .12700 | .54800 | 1.00000 |
| 22 | 1 | .22900 | .37700 | 48.00000 |
| 23 | 1 | -.00600 | .34000 | 1.00000 |
| 24 | 1 | .13500 | .24400 | 35.00000 |
| 25 | 1 | .07800 | .72300 | 1.00000 |
| 26 | 1 | .11100 | .55700 | 9.00000 |
| 27 | 1 | .18700 | .37700 | 10.00000 |
| 28 | 3 | .52700 | .99100 | 1.00000 |
| 29 | 2 | -.12000 | .88300 | 1.00000 |

SPECTRUM FROM 8-1 SPECIMEN -- TRUNCATION LEVEL 270,000 CYCLES

AKT = 4.50

RELAXATION CONSTANT C18 250000000

SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE

FLIGHT OR BLOCK NO. 1

| STMAX | STMIN | SIGFAX | STGMIN | RES | EQRES | ENN | NEP |
|-------|-------|--------|--------|--------|--------|-------|---------------|
| -1.78 | -3.98 | -3.51 | -17.92 | 0.00 | 0.00 | 1.00 | .13025831E+05 |
| 18.40 | 15.71 | 55.50 | 42.59 | -27.83 | -27.83 | 1.00 | .39342782E+03 |
| 15.71 | 12.24 | 42.89 | 27.29 | -27.83 | -15.69 | 1.00 | .56235689E+03 |
| 18.64 | 17.43 | 55.25 | 49.55 | -28.89 | -28.89 | 1.00 | .36717948E+03 |
| 19.80 | 10.63 | 55.00 | 18.20 | -29.61 | -29.61 | 1.00 | .44621680E+03 |
| 15.14 | 13.13 | 38.54 | 29.66 | -29.61 | -13.13 | 3.00 | .57667259E+03 |
| 13.69 | 6.32 | 32.20 | .87 | -29.42 | -6.61 | 1.00 | .90105168E+03 |
| 16.87 | 13.69 | 46.54 | 32.25 | -29.36 | -23.93 | 2.00 | .47906511E+03 |
| 13.69 | 11.67 | 32.34 | 23.26 | -29.28 | -6.61 | 1.00 | .71093397E+03 |
| 18.56 | 9.79 | 54.32 | 14.56 | -29.21 | -28.52 | 1.00 | .46919213E+03 |
| 9.79 | 3.58 | 14.86 | -13.10 | -29.20 | 0.00 | 1.00 | .18861779E+04 |
| 12.97 | 9.79 | 29.18 | 14.93 | -29.17 | -3.35 | 6.00 | .14033380E+04 |
| 9.79 | 8.18 | 15.32 | 8.06 | -28.74 | 3.00 | 1.00 | .42460525E+03 |
| 20.34 | 8.26 | 55.00 | .85 | -36.51 | -36.51 | 1.00 | .76807252E+03 |
| 10.18 | 8.50 | 27.28 | 1.74 | -36.51 | -8.79 | 1.00 | .58153353E+03 |
| 16.50 | 8.98 | 38.26 | 4.00 | -36.43 | -19.69 | 1.00 | .72996133E+03 |
| 13.77 | 10.79 | 25.64 | 12.18 | -36.35 | -6.98 | 7.00 | .92247844E+03 |
| 14.74 | 3.42 | 30.61 | -20.35 | -35.72 | -11.34 | 1.00 | .14935733E+04 |
| 10.14 | 6.16 | 9.97 | -7.94 | -35.66 | 3.00 | 40.00 | 4.80000000 |
| | | 10.11 | -7.81 | | | 4.00 | 4.80000000 |
| | | 10.37 | -7.55 | | | 9.60 | 4.80000000 |
| | | 10.63 | -7.29 | | | 14.40 | 4.80000000 |
| | | 10.89 | -7.03 | | | 19.20 | 4.80000000 |
| | | 11.14 | -6.77 | | | 24.00 | 4.80000000 |
| | | 11.40 | -6.52 | | | 28.80 | 4.80000000 |
| | | 11.65 | -6.27 | | | 33.60 | 4.80000000 |
| | | 11.90 | -6.02 | | | 38.40 | 4.80000000 |
| | | 12.15 | -5.77 | | | 43.20 | 4.80000000 |
| | | 12.40 | -5.52 | | | 48.00 | 4.80000000 |
| 9.15 | -1.16 | 8.04 | -33.84 | -33.12 | 0.00 | 1.00 | .30047963E+04 |
| 6.56 | 3.50 | -3.56 | -17.36 | -33.09 | 0.00 | 35.00 | .37392025E+04 |
| | | -3.52 | -17.32 | | | 3.50 | 3.50000000 |
| | | -3.45 | -17.25 | | | 7.00 | 3.50000000 |
| | | -3.38 | -17.18 | | | 10.50 | 3.50000000 |
| | | -3.31 | -17.11 | | | 14.00 | 3.50000000 |
| | | -3.24 | -17.04 | | | 17.50 | 3.50000000 |
| | | -3.17 | -16.97 | | | 21.00 | 3.50000000 |
| | | -3.10 | -16.90 | | | 24.50 | 3.50000000 |
| | | -3.03 | -16.83 | | | 28.00 | 3.50000000 |
| | | -2.96 | -16.76 | | | 31.50 | 3.50000000 |
| | | -2.89 | -16.69 | | | 35.00 | 3.50000000 |
| 19.45 | 2.10 | 55.00 | -23.08 | -32.52 | -32.52 | 1.00 | .58920931E+03 |
| 14.98 | 2.99 | 34.91 | -19.08 | -32.52 | -12.42 | 9.00 | .91708325E+03 |
| 10.14 | 5.03 | 13.57 | -9.43 | -32.07 | 0.00 | 10.00 | .16047965E+04 |
| 20.09 | 14.18 | 55.00 | 28.37 | -35.42 | -35.42 | 1.00 | .35055054E+03 |
| -7.78 | -3.98 | -38.93 | -53.34 | -35.42 | 0.00 | 1.00 | .13025831E+05 |
| 22.76 | 14.98 | 55.00 | 29.02 | -47.41 | -47.41 | 1.00 | .28748314E+03 |
| 23.07 | 13.69 | 55.00 | 15.09 | -51.52 | -51.52 | 1.00 | .27916148E+03 |
| 14.18 | 8.90 | 12.27 | -11.46 | -51.52 | -8.79 | 1.00 | .75464468E+03 |
| 16.60 | 14.18 | 12.29 | 12.40 | -51.39 | -19.69 | 19.00 | .48342514E+03 |
| | | 23.44 | 12.54 | | | 1.90 | 1.90000000 |
| | | 23.72 | 12.83 | | | 3.80 | 1.90000000 |
| | | 24.00 | 13.11 | | | 5.70 | 1.90000000 |

| | | | |
|-------|-------|--------|---------------|
| 24.60 | 13.39 | 7.03 | 1.90000000 |
| 24.56 | 13.66 | 9.50 | 1.90000000 |
| 24.83 | 13.94 | 11.40 | 1.90000000 |
| 25.11 | 14.21 | 13.30 | 1.90000000 |
| 25.37 | 14.48 | 15.20 | 1.90000000 |
| 25.64 | 14.75 | 17.10 | 1.90000000 |
| 25.91 | 15.01 | 19.00 | 1.90000000 |
| 14.18 | 12.48 | 7.52 | .65336542E+03 |
| 21.79 | 12.97 | 9.98 | .32605652E+03 |
| 18.48 | 15.63 | 34.83 | .39170979E+03 |
| 18.88 | 8.66 | 36.76 | .47468265E+03 |
| 12.38 | 10.63 | 6.23 | .90043347E+03 |
| 15.87 | 11.59 | 5.14 | .56645141E+03 |
| 14.66 | 12.40 | 9.39 | .62236611E+03 |
| 19.75 | 9.58 | 2.90 | 2.90000000 |
| 20.12 | 9.95 | 5.80 | 2.90000000 |
| 20.50 | 10.33 | 8.70 | 2.90000000 |
| 20.86 | 10.73 | 11.60 | 2.90000000 |
| 21.23 | 11.06 | 14.50 | 2.90000000 |
| 21.59 | 11.42 | 17.40 | 2.90000000 |
| 21.94 | 11.78 | 20.30 | 2.90000000 |
| 22.30 | 12.13 | 23.20 | 2.90000000 |
| 22.65 | 12.48 | 26.10 | 2.90000000 |
| 22.99 | 12.82 | 29.00 | 2.90000000 |
| -1.78 | -3.98 | -40.60 | .13025831E+05 |
| -1.05 | -3.66 | -41.80 | .14337171E+05 |

THE NUMBER OF PEAKS OR VALLEYS IN THE INPUT LOAD SPECTRUM = 75

| STEP | MAXIMUM | SIGMA | MINIMUM | COUNTER K |
|------|--------------|-------|--------------|-----------|
| 1 | -.351945E+01 | | -.179154E+02 | 1.00000 |
| 2 | .550000E+02 | | .428950E+02 | 1.00000 |
| 3 | .428950E+02 | | .272795E+02 | 1.00000 |
| 4 | .550000E+02 | | .495527E+02 | 1.00000 |
| 5 | .550000E+02 | | .182008E+02 | 1.00000 |
| 6 | .385372E+02 | | .294585E+02 | 3.00000 |
| 7 | .321966E+02 | | -.971139E+00 | 1.00000 |
| 8 | .465387E+02 | | .322548E+02 | 2.00000 |
| 9 | .323357E+02 | | .232579E+02 | 1.00000 |
| 10 | .543191E+02 | | .148568E+02 | 1.00000 |
| 11 | .148601E+02 | | -.131024E+02 | 1.00000 |
| 12 | .291797E+02 | | .148958E+02 | 6.00000 |
| 13 | .153186E+02 | | .805562E+01 | 1.00000 |
| 14 | .550000E+02 | | .648550E+00 | 1.00000 |
| 15 | .272796E+02 | | .173800E+01 | 1.00000 |
| 16 | .382570E+02 | | .399989E+01 | 1.00000 |
| 17 | .256130E+02 | | .121764E+02 | 7.00000 |
| 18 | .306127E+02 | | -.203494E+02 | 1.00000 |
| 19 | .101054E+02 | | -.781000E+01 | 4.80000 |
| 20 | .103674E+02 | | .754800E+01 | 4.80000 |
| 21 | .106275E+02 | | -.728793E+01 | 4.80000 |
| 22 | .108856E+02 | | -.702978E+01 | 4.80000 |
| 23 | .111419E+02 | | -.677353E+01 | 4.80000 |
| 24 | .113962E+02 | | -.651917E+01 | 4.80000 |
| 25 | .116487E+02 | | -.626669E+01 | 4.80000 |
| 26 | .118993E+02 | | -.601667E+01 | 4.80000 |
| 27 | .121481E+02 | | -.576729E+01 | 4.80000 |
| 28 | .123950E+02 | | -.552036E+01 | 4.80000 |
| 29 | .803921E+01 | | -.338441E+02 | 1.00000 |
| 30 | -.352359E+01 | | -.173203E+02 | 3.50000 |
| 31 | -.344941E+01 | | -.172491E+02 | 3.50000 |
| 32 | -.337838E+01 | | -.171781E+02 | 3.50000 |
| 33 | -.331750E+01 | | -.171071E+02 | 3.50000 |

| | | | |
|----|--------------|-------------|---------|
| 34 | ..323678E+01 | ..17369E+02 | 3.50000 |
| 35 | ..316621E+01 | ..16959E+02 | 3.50000 |
| 36 | ..309579E+01 | ..16855E+02 | 3.50000 |
| 37 | ..302552E+01 | ..16822E+02 | 3.50000 |
| 38 | ..295540E+01 | ..16755E+02 | 3.50000 |
| 39 | ..288544E+01 | ..16651E+02 | 3.50000 |
| 40 | ..281548E+01 | ..16547E+02 | 3.50000 |
| 41 | ..274552E+01 | ..16443E+02 | 3.50000 |
| 42 | ..267556E+01 | ..16339E+02 | 3.50000 |
| 43 | ..260560E+01 | ..16235E+02 | 3.50000 |
| 44 | ..253564E+01 | ..16131E+02 | 3.50000 |
| 45 | ..246568E+01 | ..16027E+02 | 3.50000 |
| 46 | ..239572E+01 | ..15923E+02 | 3.50000 |
| 47 | ..232576E+01 | ..15819E+02 | 3.50000 |
| 48 | ..225580E+01 | ..15715E+02 | 3.50000 |
| 49 | ..218584E+01 | ..15611E+02 | 3.50000 |
| 50 | ..211588E+01 | ..15507E+02 | 3.50000 |
| 51 | ..204592E+01 | ..15403E+02 | 3.50000 |
| 52 | ..197596E+01 | ..15299E+02 | 3.50000 |
| 53 | ..190600E+01 | ..15195E+02 | 3.50000 |
| 54 | ..183604E+01 | ..15091E+02 | 3.50000 |
| 55 | ..176608E+01 | ..14987E+02 | 3.50000 |
| 56 | ..169612E+01 | ..14883E+02 | 3.50000 |
| 57 | ..162616E+01 | ..14779E+02 | 3.50000 |
| 58 | ..155620E+01 | ..14675E+02 | 3.50000 |
| 59 | ..148624E+01 | ..14571E+02 | 3.50000 |
| 60 | ..141628E+01 | ..14467E+02 | 3.50000 |
| 61 | ..134632E+01 | ..14363E+02 | 3.50000 |
| 62 | ..127636E+01 | ..14259E+02 | 3.50000 |
| 63 | ..120640E+01 | ..14155E+02 | 3.50000 |
| 64 | ..113644E+01 | ..14051E+02 | 3.50000 |
| 65 | ..106648E+01 | ..13947E+02 | 3.50000 |
| 66 | ..099652E+01 | ..13843E+02 | 3.50000 |
| 67 | ..092656E+01 | ..13739E+02 | 3.50000 |
| 68 | ..085660E+01 | ..13635E+02 | 3.50000 |
| 69 | ..078664E+01 | ..13531E+02 | 3.50000 |
| 70 | ..071668E+01 | ..13427E+02 | 3.50000 |
| 71 | ..064672E+01 | ..13323E+02 | 3.50000 |
| 72 | ..057676E+01 | ..13219E+02 | 3.50000 |
| 73 | ..050680E+01 | ..13115E+02 | 3.50000 |
| 74 | ..043684E+01 | ..13011E+02 | 3.50000 |
| 75 | ..036688E+01 | ..12907E+02 | 3.50000 |

RANGE PAIR CYCLE COUNTED SPECTRUM

| STEP | SIGMA | MAXIMUM | MINIMUM | COUNTER K |
|------|-------|--------------|--------------|-----------|
| 1 | | -.351345E+01 | -.179154E+02 | 1.00000 |
| 2 | | .553000E+02 | .272795E+02 | 1.00000 |
| 3 | | .550030E+02 | .495527E+02 | 1.00000 |
| 4 | | .553030E+02 | -.550300E+02 | 1.00000 |
| 5 | | .321966E+02 | .294585E+02 | 1.00000 |
| 6 | | .465387E+02 | .322548E+02 | 1.00000 |
| 7 | | .465387E+02 | .232570E+02 | 1.00000 |
| 8 | | .323357E+02 | .322548E+02 | 1.00000 |
| 9 | | .543191E+02 | -.971139E+00 | 1.00000 |
| 10 | | .148601E+02 | .148568E+02 | 1.00000 |
| 11 | | .291797E+02 | .148958E+02 | 5.00000 |
| 12 | | .291797E+02 | .805562E+01 | 1.00000 |
| 13 | | .153186E+02 | .148958E+02 | 1.00000 |
| 14 | | .553000E+02 | -.131024E+02 | 1.00000 |
| 15 | | .272796E+02 | .173800E+01 | 1.00000 |
| 16 | | .362570E+02 | .648550E+00 | 1.00000 |
| 17 | | .306127E+02 | .121764E+02 | 1.00000 |
| 18 | | .101354E+02 | -.781003E+01 | 5.00000 |
| 19 | | .103674E+02 | -.754800E+01 | 5.00000 |
| 20 | | .156275E+02 | -.728793E+01 | 5.00000 |
| 21 | | .138856E+02 | -.702978E+01 | 5.00000 |
| 22 | | .111419E+02 | -.677353E+01 | 5.00000 |
| 23 | | .113962E+02 | -.651917E+01 | 5.00000 |
| 24 | | .116487E+02 | -.626659E+01 | 5.00000 |
| 25 | | .118993E+02 | -.601607E+01 | 5.00000 |
| 26 | | .121481E+02 | -.576729E+01 | 5.00000 |
| 27 | | .123950E+02 | -.552036E+01 | 4.00000 |
| 28 | | .123950E+02 | -.263494E+02 | 1.00000 |
| 29 | | .863921E+01 | -.552036E+01 | 1.00000 |
| 30 | | .352059E+01 | -.173203E+02 | 4.00000 |
| 31 | | -.344941E+01 | -.172491E+02 | 4.00000 |
| 32 | | -.337838E+01 | -.171781E+02 | 4.00000 |
| 33 | | -.330750E+01 | -.171072E+02 | 4.00000 |
| 34 | | -.323678E+01 | -.170365E+02 | 4.00000 |
| 35 | | -.316621E+01 | -.169659E+02 | 4.00000 |
| 36 | | -.309579E+01 | -.168955E+02 | 4.00000 |
| 37 | | -.302552E+01 | -.168252E+02 | 4.00000 |
| 38 | | -.295540E+01 | -.167551E+02 | 4.00000 |
| 39 | | -.288544E+01 | -.166851E+02 | 4.00000 |
| 40 | | .550000E+02 | -.190826E+02 | 1.00000 |
| 41 | | .135658E+02 | -.943374E+01 | 10.00000 |
| 42 | | .550300E+02 | -.338441E+02 | 1.00000 |
| 43 | | .550000E+02 | .200165E+02 | 1.00000 |
| 44 | | .550000E+02 | -.533997E+02 | 1.00000 |
| 45 | | .234369E+02 | .125424E+02 | 2.00000 |
| 46 | | .237213E+02 | .128688E+02 | 2.00000 |
| 47 | | .240031E+02 | .131086E+02 | 2.00000 |
| 48 | | .242824E+02 | .133079E+02 | 2.00000 |
| 49 | | .245592E+02 | .136477E+02 | 2.00000 |
| 50 | | .248335E+02 | .139390E+02 | 2.00000 |
| 51 | | .251053E+02 | .142108E+02 | 2.00000 |
| 52 | | .253747E+02 | .144822E+02 | 2.00000 |
| 53 | | .256416E+02 | .147471E+02 | 2.00000 |
| 54 | | .259062E+02 | .150117E+02 | 1.00000 |
| 55 | | .259062E+02 | .751723E+01 | 1.00000 |
| 56 | | .151434E+02 | .150117E+02 | 1.00000 |
| 57 | | .151434E+02 | .751723E+01 | 1.00000 |

| | | | |
|----|-------------|-------------|---------|
| 50 | .496632E+02 | .114204E+02 | 1.00000 |
| 51 | .268265E+02 | .219753E+02 | 1.00000 |
| 52 | .367624E+02 | .997609E+01 | 1.00000 |
| 53 | .243685E+02 | .514165E+01 | 4.00000 |
| 54 | .343686E+02 | .31136E+00 | 1.00000 |
| 55 | .197465E+02 | .997830E+01 | 3.00000 |
| 56 | .201228E+02 | .995499E+01 | 3.00000 |
| 57 | .264951E+02 | .103269E+02 | 3.00000 |
| 58 | .298634E+02 | .109522E+02 | 3.00000 |
| 59 | .212277E+02 | .110595E+02 | 3.00000 |
| 60 | .215882E+02 | .114200E+02 | 3.00000 |
| 61 | .219448E+02 | .117765E+02 | 3.00000 |
| 62 | .222977E+02 | .121295E+02 | 3.00000 |
| 63 | .226467E+02 | .124785E+02 | 3.00000 |
| 64 | .229920E+02 | .128238E+02 | 2.00000 |
| 65 | .229920E+02 | .514165E+01 | 1.00000 |
| 66 | .229920E+02 | .535478E+02 | 1.00000 |

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

| STEP | PLASTIC STRAIN | MAX OR MIN | DAMAGE |
|------|----------------|------------|-------------|
| 2 | .0696 | MAX | .59085E-03 |
| 4 | .0035 | MAX | .236703E-05 |
| 5 | .00722 | MAX | .105771E-05 |
| 14 | .09249 | MAX | .891548E-04 |
| 22 | .0304 | MAX | .49822E-07 |
| 25 | .00123 | MAX | .242990E-04 |
| 27 | .00423 | MAX | .233133E-03 |
| 28 | .00156 | MAX | .379281E-04 |
| 30 | .00053 | MIN | .718916E-05 |

DAMAGE FROM PLASTIC STRAINS= .98426328E-03

| SIGMAX | SIGMIN | RNCYC | CYCLES | ENN/CYC |
|--------|--------|-------|----------------|---------------|
| 3.51 | -17.92 | 1. | .1000000E+10 | .1000000E-08 |
| 55.00 | 27.28 | 1. | .1480786E+06 | .6753172E-05 |
| 55.00 | 49.55 | 1. | .6990694E+09 | .14306010E-08 |
| 55.00 | -55.00 | 1. | .1000000E+05 | .1000000E-03 |
| 32.22 | 29.46 | 1. | .4690198E+12 | .21321090E-11 |
| 46.54 | 32.25 | 1. | .1153687E+09 | .86678619E-08 |
| 46.54 | 23.26 | 1. | .51785551E+07 | .19310406E-06 |
| 32.34 | 32.25 | 1. | .36726287E+13 | .27228454E-12 |
| 54.32 | -.97 | 1. | .14931857E+05 | .66970905E-04 |
| 14.86 | 14.86 | 1. | .50066446E+12 | .19973469E-11 |
| 29.18 | 14.90 | 5. | .12665797E+10 | .39476395E-08 |
| 29.18 | 8.96 | 1. | .14052164E+09 | .71163418E-08 |
| 15.32 | 14.90 | 1. | .42187482E+12 | .23703714E-11 |
| 55.00 | -13.11 | 1. | .1000000E+05 | .1000000E-03 |
| 27.28 | 1.74 | 1. | .5300468E+08 | .18866263E-07 |
| 38.26 | .65 | 1. | .1745408E+07 | .57293429E-06 |
| 30.61 | 12.18 | 1. | .27966528E+09 | .35757031E-08 |
| 10.11 | -7.91 | 5. | .1000000E+10 | .5000000E-08 |
| 10.37 | -7.55 | 5. | .1000000E+10 | .5000000E-08 |
| 10.63 | -7.29 | 5. | .1000000E+10 | .5000000E-08 |
| 10.89 | -7.03 | 5. | .1000000E+10 | .5000000E-08 |
| 11.14 | -6.77 | 5. | .56054816E+09 | .89198403E-08 |
| 11.40 | -6.52 | 5. | .56088584E+09 | .89144700E-08 |
| 11.65 | -6.27 | 5. | .56115089E+09 | .89102593E-08 |
| 11.92 | -6.02 | 5. | .56134367E+09 | .89071933E-08 |
| 12.15 | -5.77 | 5. | .56144656E+09 | .89052816E-08 |
| 12.40 | -5.52 | 4. | .561511397E+09 | .71235984E-08 |
| 12.40 | -21.35 | 1. | .27376289E+08 | .36527961E-07 |
| 8.54 | -5.52 | 1. | .1000000E+10 | .1000000E-08 |
| -3.52 | -17.32 | 4. | .1000000E+10 | .4600000E-08 |
| -3.45 | -17.25 | 4. | .1000000E+10 | .4600000E-08 |
| -3.38 | -17.18 | 4. | .1000000E+10 | .4600000E-08 |

| | | | | |
|--------|--------|-----|---------------|---------------|
| -3.51 | -17.11 | 4. | .10000000E+10 | .40000000E-08 |
| -3.24 | -17.04 | 4. | .10000000E+10 | .40000000E-08 |
| -3.17 | -16.97 | 4. | .10000000E+10 | .40000000E-08 |
| -3.10 | -16.90 | 4. | .10000000E+10 | .40000000E-08 |
| -3.03 | -16.83 | 4. | .10000000E+10 | .40000000E-08 |
| -2.96 | -16.76 | 4. | .10000000E+10 | .40000000E-08 |
| -2.89 | -16.69 | 4. | .10000000E+10 | .40000000E-08 |
| 55.00 | -19.08 | 1. | .10000000E+05 | .10000000E-03 |
| 13.57 | -9.43 | 10. | .16693817E+09 | .60227117E-07 |
| 55.00 | -33.84 | 1. | .10000000E+05 | .10000000E-03 |
| 55.00 | -20.02 | 1. | .63545690E+05 | .15736461E-04 |
| 55.00 | -53.34 | 1. | .10000000E+05 | .10000000E-03 |
| 23.44 | 12.54 | 2. | .52786328E+10 | .37888599E-05 |
| 23.72 | 12.83 | 2. | .52805280E+10 | .37875000E-05 |
| 24.00 | 13.11 | 2. | .52807566E+10 | .37873361E-05 |
| 24.28 | 13.39 | 2. | .52793259E+10 | .37883625E-05 |
| 24.56 | 13.66 | 2. | .52762450E+10 | .37935745E-05 |
| 24.83 | 13.94 | 2. | .52715248E+10 | .37935687E-05 |
| 25.11 | 14.21 | 2. | .52651779E+10 | .37985421E-05 |
| 25.37 | 14.48 | 2. | .52572184E+10 | .38042931E-05 |
| 25.64 | 14.75 | 2. | .52476624E+10 | .38112208E-05 |
| 25.91 | 15.01 | 1. | .52365271E+10 | .19096626E-05 |
| 25.91 | 7.52 | 1. | .37589154E+09 | .26603419E-08 |
| 15.14 | 15.01 | 1. | .48307951E+12 | .20700526E-11 |
| 15.14 | 7.52 | 1. | .15515977E+11 | .64449697E-10 |
| 49.68 | -11.46 | 1. | .24152055E+05 | .41404344E-04 |
| 34.83 | -22.00 | 1. | .15814258E+10 | .63234078E-09 |
| 36.76 | 9.98 | 1. | .15038306E+08 | .66496853E-07 |
| 24.39 | 5.14 | 4. | .31310615E+09 | .12775220E-07 |
| 24.39 | -3.1 | 1. | .77428372E+08 | .12915162E-07 |
| 19.75 | 9.58 | 3. | .67337984E+10 | .44551379E-05 |
| 20.12 | 9.95 | 3. | .67798754E+10 | .44248601E-09 |
| 20.50 | 10.33 | 3. | .68230691E+10 | .43968484E-09 |
| 20.86 | 10.70 | 3. | .68633296E+10 | .43710563E-09 |
| 21.23 | 11.06 | 3. | .69036119E+10 | .43474405E-09 |
| 21.59 | 11.42 | 3. | .69348755E+10 | .43259609E-09 |
| 21.94 | 11.78 | 3. | .69660846E+10 | .43065799E-09 |
| 22.30 | 12.13 | 3. | .69942085E+10 | .42892630E-09 |
| 22.65 | 12.48 | 3. | .70192214E+10 | .42739783E-09 |
| 22.99 | 12.82 | 2. | .70411021E+10 | .28434644E-09 |
| 22.99 | 5.14 | 1. | .49378912E+09 | .20251560E-08 |
| -41.80 | -53.54 | 1. | .10000000E+10 | .10000000E-08 |

DAMAGE PER THIS SET= .16162546E-02

TOTAL ENN/CYC =, .16162546E-02

FLIGHT OR-BLOCK NO. 2

| STMAX | STMIN | SIGMAX | SIGMIN | RES | EORES | ENN | NEP |
|-------|-------|--------|--------|--------|--------|------|---------------|
| -7.8 | -3.98 | -40.58 | -54.99 | -37.67 | 0.01 | 1.00 | .13028031E+05 |
| 18.40 | 15.71 | 45.73 | 33.63 | -37.07 | -27.81 | 1.00 | .39342782E+03 |
| 15.71 | 12.24 | 33.68 | 18.37 | -37.01 | -15.69 | 1.00 | .56235689E+03 |
| 18.64 | 17.43 | 46.96 | 41.52 | -36.92 | -28.89 | 1.00 | .36717948E+03 |
| 18.80 | 10.63 | 47.74 | 13.94 | -36.87 | -29.61 | 1.00 | .44621680E+03 |
| 15.14 | 13.13 | 31.31 | 22.24 | -35.84 | -13.15 | 3.00 | .57667259E+03 |
| 13.69 | 6.32 | 25.06 | 8.11 | -36.55 | -6.61 | 1.00 | .93105168E+03 |
| 16.87 | 13.69 | 39.42 | 25.14 | -36.48 | -20.91 | 2.00 | .47906511E+03 |
| 13.69 | 11.67 | 25.29 | 16.21 | -36.33 | -6.61 | 1.00 | .71099397E+03 |
| 18.56 | 9.79 | 47.29 | 7.83 | -36.23 | -28.52 | 1.00 | .46919213E+03 |
| 9.79 | 3.58 | 7.87 | -23.13 | -36.23 | 0.03 | 1.00 | .18861779E+04 |
| 12.97 | 9.79 | 22.20 | 7.91 | -35.15 | -3.35 | 6.00 | .83680259E+03 |
| 9.79 | 8.18 | 8.45 | 1.19 | -35.61 | 0.01 | 1.00 | .14033389E+04 |
| 20.54 | 8.26 | 55.00 | .65 | -36.51 | -36.51 | 1.00 | .42460525E+03 |
| 14.18 | 8.50 | 27.28 | 1.74 | -36.51 | -8.79 | 1.00 | .76807552E+03 |
| 16.60 | 8.98 | 38.26 | 4.93 | -36.43 | -19.69 | 1.00 | .58153350E+03 |
| 13.77 | 10.79 | 25.61 | 12.18 | -36.36 | -6.98 | 7.00 | .72996103E+03 |
| 14.74 | 3.42 | 30.61 | -20.35 | -35.72 | -11.33 | 1.00 | .92247814E+03 |

| | | | | | | | | |
|-------|-------|--------|--------|--------|--------|---------------|----|------------|
| 10.14 | 0.16 | 9.97 | -7.34 | -35.60 | 4.00 | 1.493573E+04 | 19 | 4.80000000 |
| | | 11.11 | 7.81 | | 4.80 | | | 4.80000000 |
| | | 11.37 | 7.73 | | 4.80 | | | 4.80000000 |
| | | 11.53 | 7.69 | | 14.40 | | | 4.80000000 |
| | | 11.83 | 7.33 | | 18.20 | | | 4.80000000 |
| | | 13.14 | 6.77 | | 24.00 | | | 4.80000000 |
| | | 11.40 | 6.52 | | 28.80 | | | 4.80000000 |
| | | 11.65 | 6.27 | | 33.60 | | | 4.80000000 |
| | | 11.90 | 6.02 | | 38.40 | | | 4.80000000 |
| | | 12.15 | 5.77 | | 43.20 | | | 4.80000000 |
| | | 12.40 | 5.52 | | 48.00 | | | 4.80000000 |
| 9.15 | -0.16 | 8.04 | -33.84 | -33.12 | 0.00 | .30647963E+04 | 20 | 4.80000000 |
| 6.56 | 3.50 | -3.56 | -17.36 | -33.09 | 0.00 | .37392025E+04 | 21 | 3.50000000 |
| | | -3.52 | -17.32 | | 3.50 | | | 3.50000000 |
| | | -3.45 | -17.25 | | 7.00 | | | 3.50000000 |
| | | -3.38 | -17.18 | | 10.50 | | | 3.50000000 |
| | | -3.31 | -17.11 | | 14.00 | | | 3.50000000 |
| | | -3.24 | -17.04 | | 17.50 | | | 3.50000000 |
| | | -3.17 | -16.97 | | 21.00 | | | 3.50000000 |
| | | -3.10 | -16.90 | | 24.50 | | | 3.50000000 |
| | | -3.03 | -16.83 | | 28.00 | | | 3.50000000 |
| | | -2.96 | -16.76 | | 31.50 | | | 3.50000000 |
| | | -2.89 | -16.69 | | 35.00 | | | 3.50000000 |
| 19.45 | 2.18 | 55.00 | -23.38 | -32.52 | -32.52 | .58923931E+03 | 22 | 1.90000000 |
| 14.98 | 2.99 | 37.91 | -19.08 | -32.52 | 9.00 | .91708325E+03 | 23 | 1.90000000 |
| 10.14 | 5.03 | 13.57 | -9.43 | -32.17 | 0.00 | .16647965E+04 | 24 | 1.90000000 |
| 20.79 | 14.18 | 55.00 | 28.37 | -35.42 | 1.00 | .35855054E+03 | 25 | 1.90000000 |
| -7.8 | -3.98 | -38.93 | -53.34 | -35.42 | 0.00 | .13225831E+05 | 26 | 1.90000000 |
| 22.76 | 14.98 | 55.00 | 28.37 | -47.41 | 1.00 | .28748314E+03 | 27 | 1.90000000 |
| 23.67 | 13.69 | 55.00 | 14.09 | -51.52 | 1.00 | .27916148E+03 | 28 | 1.90000000 |
| 14.18 | 8.90 | 12.27 | -11.46 | -51.52 | 1.00 | .75464463E+03 | 29 | 1.90000000 |
| 16.60 | 14.18 | 23.29 | 12.43 | -51.39 | 19.00 | .48342514E+03 | 30 | 1.90000000 |
| | | 23.44 | 12.54 | | 1.00 | | | 1.90000000 |
| | | 23.72 | 12.83 | | 3.88 | | | 1.90000000 |
| | | 24.00 | 13.11 | | 5.70 | | | 1.90000000 |
| | | 24.28 | 13.39 | | 7.60 | | | 1.90000000 |
| | | 24.56 | 13.66 | | 9.50 | | | 1.90000000 |
| | | 24.83 | 13.94 | | 11.40 | | | 1.90000000 |
| | | 25.11 | 14.21 | | 13.30 | | | 1.90000000 |
| | | 25.37 | 14.48 | | 15.20 | | | 1.90000000 |
| | | 25.64 | 14.75 | | 17.10 | | | 1.90000000 |
| | | 25.91 | 15.01 | | 19.00 | | | 1.90000000 |
| 14.18 | 12.48 | 15.14 | 7.52 | -48.65 | -8.79 | .65336542E+03 | 31 | 2.90000000 |
| 21.79 | 12.97 | 49.68 | 9.98 | -48.37 | -43.05 | .32605652E+03 | 32 | 2.90000000 |
| 16.48 | 15.63 | 34.83 | 22.00 | -48.33 | -28.15 | .39170979E+03 | 33 | 2.90000000 |
| 18.88 | 8.66 | 36.76 | -9.24 | -48.21 | -29.98 | .47468264E+03 | 34 | 2.90000000 |
| 12.38 | 10.63 | 6.23 | -3.31 | -48.13 | 0.00 | .90043347E+03 | 35 | 2.90000000 |
| 15.87 | 11.59 | 24.39 | 5.14 | -47.03 | -16.42 | .56645141E+03 | 36 | 2.90000000 |
| 14.66 | 12.40 | 19.56 | 9.39 | -46.41 | -10.97 | .62236611E+03 | 37 | 2.90000000 |
| | | 19.75 | 9.58 | | 2.90 | | | 2.90000000 |
| | | 20.12 | 9.95 | | 5.80 | | | 2.90000000 |
| | | 20.50 | 10.33 | | 8.70 | | | 2.90000000 |
| | | 20.86 | 10.70 | | 11.60 | | | 2.90000000 |
| | | 21.23 | 11.06 | | 14.50 | | | 2.90000000 |
| | | 21.59 | 11.42 | | 17.40 | | | 2.90000000 |
| | | 21.94 | 11.78 | | 20.30 | | | 2.90000000 |
| | | 22.30 | 12.13 | | 23.20 | | | 2.90000000 |
| | | 22.65 | 12.48 | | 26.10 | | | 2.90000000 |
| | | 22.99 | 12.82 | | 29.00 | | | 2.90000000 |
| -7.8 | -3.98 | -40.60 | -55.00 | -37.08 | 0.00 | .13025031E+05 | 38 | 2.90000000 |
| -1.05 | -3.66 | -41.80 | -53.54 | -37.08 | 0.00 | .14337171E+05 | 39 | 2.90000000 |

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

14 .00032
22 .00004
25 .00123
27 .00420
28 .00156
38 .00063

MAX
MAX
MAX
MAX
MAX
MIN

.202856E-05
.498022E-07
.242990E-04
.233133E-03
.379281E-04
.718916E-05
DAMAGE FROM PLASTIC STRAINS= .30462764E-03

| SIGMAX | SIGMIN | RNCYC | CYCLES | ENN/CYC |
|--------|--------|-------|---------------|---------------|
| -40.58 | -54.99 | 1. | .10000000E+10 | .10000000E-08 |
| 45.73 | 18.07 | 1. | .24093777E+07 | .41504493E-06 |
| 33.68 | 33.63 | 1. | .43206072E+13 | .23144898E-12 |
| 46.96 | 41.52 | 1. | .23661145E+11 | .42263381E-10 |
| 47.74 | -20.13 | 1. | .19780666E+05 | .50554416E-04 |
| 25.06 | 22.24 | 1. | .27937764E+12 | .35793846E-11 |
| 39.42 | 25.14 | 1. | .59516846E+09 | .19795377E-08 |
| 39.42 | 16.21 | 1. | .24432813E+08 | .40928567E-07 |
| 25.29 | 25.14 | 1. | .15753982E+13 | .63476016E-12 |
| 47.29 | -8.11 | 1. | .58801786E+05 | .17006286E-04 |
| 7.87 | 7.83 | 1. | .10000000E+10 | .10000000E-08 |
| 22.20 | 7.91 | 5. | .15461310E+10 | .32338787E-08 |
| 22.20 | 1.19 | 1. | .21206560E+09 | .47155219E-08 |
| 8.45 | 7.91 | 1. | .10000000E+10 | .10000000E-08 |
| 55.00 | -53.34 | 1. | .10000000E+10 | .10000000E-08 |
| 27.28 | 1.74 | 1. | .10000000E+05 | .10000000E-03 |
| 38.26 | .65 | 1. | .53004688E+08 | .18866263E-07 |
| 30.61 | 12.18 | 1. | .17454308E+07 | .57293429E-06 |
| 10.11 | -7.81 | 1. | .27966528E+09 | .35757031E-08 |
| 10.37 | -7.55 | 5. | .10000000E+10 | .50000000E-08 |
| 10.63 | -7.29 | 5. | .10000000E+10 | .50000000E-08 |
| 10.89 | -7.03 | 5. | .10000000E+10 | .50000000E-08 |
| 11.14 | -6.77 | 5. | .10000000E+10 | .50000000E-08 |
| 11.40 | -6.52 | 5. | .56054816E+09 | .89198403E-08 |
| 11.65 | -6.27 | 5. | .56388584E+09 | .89144700E-08 |
| 11.90 | -6.02 | 5. | .56115089E+09 | .89102593E-08 |
| 12.15 | -5.77 | 5. | .56134367E+09 | .89071993E-08 |
| 12.40 | -5.52 | 4. | .56146456E+09 | .89052816E-08 |
| 12.60 | -5.27 | 4. | .56151397E+09 | .71235984E-08 |
| 12.80 | -5.02 | 1. | .27376289E+08 | .36527961E-07 |
| 13.00 | -4.77 | 1. | .10000000E+10 | .10000000E-08 |
| 13.20 | -4.52 | 4. | .10000000E+10 | .10000000E-08 |
| 13.40 | -4.27 | 4. | .10000000E+10 | .10000000E-08 |
| 13.60 | -4.02 | 4. | .10000000E+10 | .10000000E-08 |
| 13.80 | -3.77 | 4. | .10000000E+10 | .10000000E-08 |
| 14.00 | -3.52 | 4. | .10000000E+10 | .10000000E-08 |
| 14.20 | -3.27 | 4. | .10000000E+10 | .10000000E-08 |
| 14.40 | -3.02 | 4. | .10000000E+10 | .10000000E-08 |
| 14.60 | -2.77 | 4. | .10000000E+10 | .10000000E-08 |
| 14.80 | -2.52 | 4. | .10000000E+10 | .10000000E-08 |
| 15.00 | -2.27 | 4. | .10000000E+10 | .10000000E-08 |
| 15.20 | -2.02 | 4. | .10000000E+10 | .10000000E-08 |
| 15.40 | -1.77 | 4. | .10000000E+10 | .10000000E-08 |
| 15.60 | -1.52 | 4. | .10000000E+10 | .10000000E-08 |
| 15.80 | -1.27 | 4. | .10000000E+10 | .10000000E-08 |
| 16.00 | -1.02 | 4. | .10000000E+10 | .10000000E-08 |
| 16.20 | -0.77 | 4. | .10000000E+10 | .10000000E-08 |
| 16.40 | -0.52 | 4. | .10000000E+10 | .10000000E-08 |
| 16.60 | -0.27 | 4. | .10000000E+10 | .10000000E-08 |
| 16.80 | 0.02 | 4. | .10000000E+10 | .10000000E-08 |
| 17.00 | 0.27 | 4. | .10000000E+10 | .10000000E-08 |
| 17.20 | 0.52 | 4. | .10000000E+10 | .10000000E-08 |
| 17.40 | 0.77 | 4. | .10000000E+10 | .10000000E-08 |
| 17.60 | 1.02 | 4. | .10000000E+10 | .10000000E-08 |
| 17.80 | 1.27 | 4. | .10000000E+10 | .10000000E-08 |
| 18.00 | 1.52 | 4. | .10000000E+10 | .10000000E-08 |
| 18.20 | 1.77 | 4. | .10000000E+10 | .10000000E-08 |
| 18.40 | 2.02 | 4. | .10000000E+10 | .10000000E-08 |
| 18.60 | 2.27 | 4. | .10000000E+10 | .10000000E-08 |
| 18.80 | 2.52 | 4. | .10000000E+10 | .10000000E-08 |
| 19.00 | 2.77 | 4. | .10000000E+10 | .10000000E-08 |
| 19.20 | 3.02 | 4. | .10000000E+10 | .10000000E-08 |
| 19.40 | 3.27 | 4. | .10000000E+10 | .10000000E-08 |
| 19.60 | 3.52 | 4. | .10000000E+10 | .10000000E-08 |
| 19.80 | 3.77 | 4. | .10000000E+10 | .10000000E-08 |
| 20.00 | 4.02 | 4. | .10000000E+10 | .10000000E-08 |
| 20.20 | 4.27 | 4. | .10000000E+10 | .10000000E-08 |
| 20.40 | 4.52 | 4. | .10000000E+10 | .10000000E-08 |
| 20.60 | 4.77 | 4. | .10000000E+10 | .10000000E-08 |
| 20.80 | 5.02 | 4. | .10000000E+10 | .10000000E-08 |
| 21.00 | 5.27 | 4. | .10000000E+10 | .10000000E-08 |
| 21.20 | 5.52 | 4. | .10000000E+10 | .10000000E-08 |
| 21.40 | 5.77 | 4. | .10000000E+10 | .10000000E-08 |
| 21.60 | 6.02 | 4. | .10000000E+10 | .10000000E-08 |
| 21.80 | 6.27 | 4. | .10000000E+10 | .10000000E-08 |
| 22.00 | 6.52 | 4. | .10000000E+10 | .10000000E-08 |
| 22.20 | 6.77 | 4. | .10000000E+10 | .10000000E-08 |
| 22.40 | 7.02 | 4. | .10000000E+10 | .10000000E-08 |
| 22.60 | 7.27 | 4. | .10000000E+10 | .10000000E-08 |
| 22.80 | 7.52 | 4. | .10000000E+10 | .10000000E-08 |
| 23.00 | 7.77 | 4. | .10000000E+10 | .10000000E-08 |
| 23.20 | 8.02 | 4. | .10000000E+10 | .10000000E-08 |
| 23.40 | 8.27 | 4. | .10000000E+10 | .10000000E-08 |
| 23.60 | 8.52 | 4. | .10000000E+10 | .10000000E-08 |
| 23.80 | 8.77 | 4. | .10000000E+10 | .10000000E-08 |
| 24.00 | 9.02 | 4. | .10000000E+10 | .10000000E-08 |
| 24.20 | 9.27 | 4. | .10000000E+10 | .10000000E-08 |
| 24.40 | 9.52 | 4. | .10000000E+10 | .10000000E-08 |
| 24.60 | 9.77 | 4. | .10000000E+10 | .10000000E-08 |
| 24.80 | 10.02 | 4. | .10000000E+10 | .10000000E-08 |
| 25.00 | 10.27 | 4. | .10000000E+10 | .10000000E-08 |
| 25.20 | 10.52 | 4. | .10000000E+10 | .10000000E-08 |
| 25.40 | 10.77 | 4. | .10000000E+10 | .10000000E-08 |
| 25.60 | 11.02 | 4. | .10000000E+10 | .10000000E-08 |
| 25.80 | 11.27 | 4. | .10000000E+10 | .10000000E-08 |
| 26.00 | 11.52 | 4. | .10000000E+10 | .10000000E-08 |
| 26.20 | 11.77 | 4. | .10000000E+10 | .10000000E-08 |
| 26.40 | 12.02 | 4. | .10000000E+10 | .10000000E-08 |
| 26.60 | 12.27 | 4. | .10000000E+10 | .10000000E-08 |
| 26.80 | 12.52 | 4. | .10000000E+10 | .10000000E-08 |
| 27.00 | 12.77 | 4. | .10000000E+10 | .10000000E-08 |
| 27.20 | 13.02 | 4. | .10000000E+10 | .10000000E-08 |
| 27.40 | 13.27 | 4. | .10000000E+10 | .10000000E-08 |
| 27.60 | 13.52 | 4. | .10000000E+10 | .10000000E-08 |
| 27.80 | 13.77 | 4. | .10000000E+10 | .10000000E-08 |
| 28.00 | 14.02 | 4. | .10000000E+10 | .10000000E-08 |
| 28.20 | 14.27 | 4. | .10000000E+10 | .10000000E-08 |
| 28.40 | 14.52 | 4. | .10000000E+10 | .10000000E-08 |
| 28.60 | 14.77 | 4. | .10000000E+10 | .10000000E-08 |
| 28.80 | 15.02 | 4. | .10000000E+10 | .10000000E-08 |
| 29.00 | 15.27 | 4. | .10000000E+10 | .10000000E-08 |
| 29.20 | 15.52 | 4. | .10000000E+10 | .10000000E-08 |
| 29.40 | 15.77 | 4. | .10000000E+10 | .10000000E-08 |
| 29.60 | 16.02 | 4. | .10000000E+10 | .10000000E-08 |
| 29.80 | 16.27 | 4. | .10000000E+10 | .10000000E-08 |
| 30.00 | 16.52 | 4. | .10000000E+10 | .10000000E-08 |
| 30.20 | 16.77 | 4. | .10000000E+10 | .10000000E-08 |
| 30.40 | 17.02 | 4. | .10000000E+10 | .10000000E-08 |
| 30.60 | 17.27 | 4. | .10000000E+10 | .10000000E-08 |
| 30.80 | 17.52 | 4. | .10000000E+10 | .10000000E-08 |
| 31.00 | 17.77 | 4. | .10000000E+10 | .10000000E-08 |
| 31.20 | 18.02 | 4. | .10000000E+10 | .10000000E-08 |
| 31.40 | 18.27 | 4. | .10000000E+10 | .10000000E-08 |
| 31.60 | 18.52 | 4. | .10000000E+10 | .10000000E-08 |
| 31.80 | 18.77 | 4. | .10000000E+10 | .10000000E-08 |
| 32.00 | 19.02 | 4. | .10000000E+10 | .10000000E-08 |
| 32.20 | 19.27 | 4. | .10000000E+10 | .10000000E-08 |
| 32.40 | 19.52 | 4. | .10000000E+10 | .10000000E-08 |
| 32.60 | 19.77 | 4. | .10000000E+10 | .10000000E-08 |
| 32.80 | 20.02 | 4. | .10000000E+10 | .10000000E-08 |
| 33.00 | 20.27 | 4. | .10000000E+10 | .10000000E-08 |
| 33.20 | 20.52 | 4. | .10000000E+10 | .10000000E-08 |
| 33.40 | 20.77 | 4. | .10000000E+10 | .10000000E-08 |
| 33.60 | 21.02 | 4. | .10000000E+10 | .10000000E-08 |
| 33.80 | 21.27 | 4. | .10000000E+10 | .10000000E-08 |
| 34.00 | 21.52 | 4. | .10000000E+10 | .10000000E-08 |
| 34.20 | 21.77 | 4. | .10000000E+10 | .10000000E-08 |
| 34.40 | 22.02 | 4. | .10000000E+10 | .10000000E-08 |
| 34.60 | 22.27 | 4. | .10000000E+10 | .10000000E-08 |
| 34.80 | 22.52 | 4. | .10000000E+10 | .10000000E-08 |
| 35.00 | 22.77 | 4. | .10000000E+10 | .10000000E-08 |
| 35.20 | 23.02 | 4. | .10000000E+10 | .10000000E-08 |
| 35.40 | 23.27 | 4. | .10000000E+10 | .10000000E-08 |
| 35.60 | 23.52 | 4. | .10000000E+10 | .10000000E-08 |
| 35.80 | 23.77 | 4. | .10000000E+10 | .10000000E-08 |
| 36.00 | 24.02 | 4. | .10000000E+10 | .10000000E-08 |
| 36.20 | 24.27 | 4. | .10000000E+10 | .10000000E-08 |
| 36.40 | 24.52 | 4. | .10000000E+10 | .10000000E-08 |
| 36.60 | 24.77 | 4. | .10000000E+10 | .10000000E-08 |
| 36.80 | 25.02 | 4. | .10000000E+10 | .10000000E-08 |
| 37.00 | 25.27 | 4. | .10000000E+10 | .10000000E-08 |
| 37.20 | 25.52 | 4. | .10000000E+10 | .10000000E-08 |
| 37.40 | 25.77 | 4. | .10000000E+10 | .10000000E-08 |
| 37.60 | 26.02 | 4. | .10000000E+10 | .10000000E-08 |
| 37.80 | 26.27 | 4. | .10000000E+10 | .10000000E-08 |
| 38.00 | 26.52 | 4. | .10000000E+10 | .10000000E-08 |
| 38.20 | 26.77 | 4. | .10000000E+10 | .10000000E-08 |
| 38.40 | 27.02 | 4. | .10000000E+10 | .10000000E-08 |
| 38.60 | 27.27 | 4. | .10000000E+10 | .10000000E-08 |
| 38.80 | 27.52 | 4. | .10000000E+10 | .10000000E-08 |
| 39.00 | 27.77 | 4. | .10000000E+10 | .10000000E-08 |
| 39.20 | 28.02 | 4. | .10000000E+10 | .10000000E-08 |
| 39.40 | 28.27 | 4. | .10000000E+10 | .10000000E-08 |
| 39.60 | 28.52 | 4. | .10000000E+10 | .10000000E-08 |
| 39.80 | 28.77 | 4. | .10000000E+10 | .10000000E-08 |
| 40.00 | 29.02 | 4. | .10000000E+10 | .10000000E-08 |
| 40.20 | 29.27 | 4. | .10000000E+10 | .10000000E-08 |
| 40.40 | 29.52 | 4. | .10000000E+10 | .10000000E-08 |
| 40.60 | 29.77 | 4. | .10000000E+10 | .10000000E-08 |
| 40.80 | 30.02 | 4. | .10000000E+10 | .10000000E-08 |
| 41.00 | 30.27 | 4. | .10000000E+10 | .10000000E-08 |
| 41.20 | 30.52 | 4. | .10000000E+10 | .10000000E-08 |
| 41.40 | 30.77 | 4. | .10000000E+10 | .10000000E-08 |
| 41.60 | 31.02 | 4. | .10000000E+10 | .10000000E-08 |
| 41.80 | 31.27 | 4. | .10000000E+10 | .10000000E-08 |
| 42.00 | 31.52 | 4. | .10000000E+10 | .10000000E-08 |
| 42.20 | 31.77 | 4. | .10000000E+10 | .10000000E-08 |
| 42.40 | 32.02 | 4. | .10000000E+10 | .10000000E-08 |
| 42.60 | 32.27 | 4. | .10000000E+10 | .10000000E-08 |
| 42.80 | 32.52 | 4. | .10000000E+10 | .10000000E-08 |
| 43.00 | 32.77 | 4. | .10000000E+10 | .10000000E-08 |
| 43.20 | 33.02 | 4. | .10000000E+10 | .10000000E-08 |
| 43.40 | 33.27 | 4. | .10000000E+10 | .10000000E-08 |
| 43.60 | 33.52 | 4. | .10000000E+10 | .10000000E-08 |
| 43.80 | 33.77 | 4. | .10000000E+10 | .10000000E-08 |
| 44.00 | 34.02 | 4. | .10000000E+10 | .10000000E-08 |
| 44.20 | 34.27 | 4. | .10000000E+10 | .10000000E-08 |
| 44.40 | 34.52 | 4. | .10000000E+10 | .10000000E-08 |
| 44.60 | 34.77 | 4. | .10000000E+10 | .10000000E-08 |
| 44.80 | 35.02 | 4. | .10000000 | |

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-15.20 -29.00

19.45 2.10 42.84 -35.24

14.98 2.99 22.79 -31.20

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-44.68 -32.52

10.14 5.03 1.72 -21.28

23.75 -3.23 60.71 -68.71

25

9.00

-44.64 -12.42

20.09 14.18 44.22 17.59

-7.6 -3.98 -40.60 -55.00

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19.00

-46.21 -35.42

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

27

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-47.41 -47.41

14.18 8.90 12.27 -11.46

16.60 14.18 23.29 12.40

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-51.52 -51.52

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

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22.76 14.98 55.00 20.02

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22.76 14.98 55.00 20.02

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22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

36

1.00

-51.39 -19.69

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

37

1.00

-51.39 -19.69

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

38

1.00

-51.39 -19.69

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

39

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-51.39 -19.69

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

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-51.39 -19.69

22.76 14.98 55.00 20.02

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23.67 13.69 55.00 16.09

42

1.00

-51.39 -19.69

22.76 14.98 55.00 20.02

23.67 13.69 55.00 16.09

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

| STEP | PLASTIC STRAIN | MAX OR MIN | DAMAGE |
|------|----------------|------------|-------------|
| 2 | .00593 | MAX | .438461E-03 |
| 27 | .00321 | MAX | .141845E-03 |
| 29 | .00185 | MIN | .183981E-04 |
| 30 | .00365 | MAX | .180144E-03 |
| 31 | .00156 | MAX | .379281E-04 |
| 41 | .00063 | MIN | .718916E-05 |

DAMAGE FROM PLASTIC STRAINS= .82396546E-03

| SIGMAX | SIGMIN | RNOYC | CYCLES | ENN/CYC |
|--------|--------|-------|---------------|---------------|
| -40.58 | -54.99 | 1. | .1000000E+10 | .1000000E-06 |
| 55.00 | -46.42 | 1. | .1000000E+05 | .1000000E-03 |
| 29.46 | 17.35 | 1. | .29397878E+10 | .34016060E-09 |
| 17.50 | 17.35 | 1. | .63000870E+12 | .15872797E-11 |
| 30.85 | 25.43 | 1. | .77590688E+11 | .12888145E-10 |
| 31.73 | 1.89 | 1. | .14305010E+08 | .69905579E-07 |
| 9.32 | 6.31 | 1. | .13000000E+10 | .1000000E-08 |
| 23.72 | 9.44 | 1. | .15092544E+10 | .66257844E-09 |
| 23.72 | .66 | 1. | .11830863E+09 | .84524691E-08 |
| 9.44 | 9.44 | 1. | .1000000E+10 | .1000000E-08 |
| 31.79 | -23.85 | 1. | .46704942E+06 | .21411011E-05 |
| -7.56 | -7.67 | 1. | .4000000E+10 | .1000000E-08 |

FLIGHT OR BLOCK NO. 4

TOTAL ENN/CYC =, .37115261E-02

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE PER THIS SET= .83070929E-03

FLIGHT OR BLOCK NO. 5

TOTAL ENN/CYC =, .45422354E-02

DAMAGE FROM PLASTIC STRAINS= .22665690E-02

DAMAGE PER THIS SET= .26509046E-02

FLIGHT OR BLOCK NO. 6

TOTAL ENN/CYC =, .71931400E-02

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE PER THIS SET= .83070929E-03

FLIGHT OR BLOCK NO. 7

TOTAL ENN/CYC =, .80238493E-02

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE PER THIS SET= .83070929E-03

TOTAL ENN/CYC =, .88545586E-02

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PCN N102001

APPENDIX III

LIST OF COMPUTER PROGRAM SYMBOLS AND DEFINITIONS

| | |
|--------|---|
| A | Coefficient of the x^2 term in the equation of a line on a constant life fatigue diagram where minimum stress is x and maximum stress is y. ($R = Ax^2 + Bx + C - y$) |
| AA | An assigned value of +1. or -1. |
| AAA | A stress used in the calculation of plastic strain. |
| ABDIF | The absolute value of DIF. |
| ABM | The absolute value of ASMAX or of ASMIN, as assigned. |
| ABMAX | The absolute value of ASMAX. |
| ABMEAN | The absolute value of ASMEAN. |
| ABMIN | The absolute value of ASMIN. |
| ABR4 | The absolute value of R(4). |
| ABR7 | The absolute value of R(7). |
| ABS | The name of a routine calling for the absolute value of a quantity. |
| AKT | Stress concentration factor, K_t |
| ASE | Array of values of ENN, for the plotting subroutine. |
| ASMAX | The product (AKT) (STMAX) |
| ASMEAN | The quantity (ASMAX + ASMIN)/2 |
| ASMIN | The product (AKT) (STMIN) |
| ASN | Array of values of ASMIN, for the plotting subroutine. |
| ASX | Array of values of ASMAX, for the plotting subroutine. |
| AVSGMN | Average value of SIGMIN over an interval. |
| AVSGMX | Average value of SIGMAX over an interval. |

B Coefficient of the x term. (See A.)

BBB A stress used in the calculation of plastic strain.

C The constant. (See A.)

COFMAN Inverse of the Coffin-Manson slope.

CYCINT The number of cycles in an interval.

CYCLES The calculated number of cycles expected to be indicated on a constant life fatigue diagram for the applied combination of maximum and minimum stress.

C1 The Residual Stress Relaxation Constant (See ENEP,)

DAM Damage.

DECK Decimal or real value of integer K after conversion.

DEL2 A portion of a least-squares-method solution.

DIF The difference between residual stress and equilibrium residual stress. (RES(I) - EQRES)

DO2 A portion of a least-squares-method solution.

DUMMY A variable used in the calculation of the number of cycles to be considered as an interval for relaxation determination.

ELMOD The elastic modulus.

EN The number of cycles from the beginning of the relaxation process to the end of the current interval.

ENEP The number of cycles required for overload residual stress effect to return to within one-tenth of its original difference from equilibrium conditions.

$$(N_{ep} = C1/(ABM)^{E1} (ABMEAN)^{E2})$$

ENN The number of applied cycles at a load level.

ENNCYC The ratio of the number of applied cycles to the number of cycles to failure. (ENN/CYCLES)

EPSD ICF strain intercept.

EQRES Equilibrium residual stress.

EX An exponential function depicting the relaxation of residual stress.

EXP The name of a routine calling for the exponential value of a quantity.

EXPO An exponent. The power of 10 which indicates the number of cycles to failure.

E1 } Residual stress Relaxation Exponents.
E2 }

EXOAT The name of a routine calling for integer-to-real conversion.

I A variable subscript.

IBLOCK The identifying number of a block the blocks being numbered consecutively from 1 to NBLOCK.

IFIX The name of a routine calling for real-to-integer conversion.

IN The number of steps input to the range pair counting subroutine.

IPRINT Value controlling the WRITE statements.

IRAIN A counter.

IRPOM Value controlling entry into the range pair counting subroutine.

| | |
|--------|--|
| ISTEP | The identifying step number, the steps being numbered from 1 to NLEVEL. |
| ITYPE | The identifying type number, the types being numbered from 1 to NTYPE. |
| J | A variable subscript. |
| JA | Value of +1 or 0, as assigned for branch determination. |
| JB | Value of -1 or 0, as assigned for branch determination. |
| JJ | An index variable. |
| JJJ | An index variable. |
| JKL | An index variable. |
| K | An index variable. |
| KK | An index variable. |
| KPMAX | The number of steps output from the range pair counting subroutine. |
| L | An index variable. |
| LMN | An index variable. |
| M | An index variable. |
| N | An index variable with values of N=4-7 indicating the power of 10, and thus identifying a particular life cycle curve. |
| NBLOCK | The total number of times to execute a block of loads. |
| NDECK | The number of data decks to be run sequentially. |
| NFLAG | An integer used as a counter. |
| NFLAG2 | An integer used as a counter. |

NLEVEL The total number of steps, or levels, of loads in a block.

NN A subscripted variable used to indicate which types of
 loads are experienced in which blocks.

NTYPE The total number of different types.

PLSTRA Plastic strain.

R Residue term in damage calculation.

RES Residual stress.

RNCYC The number of cycles for a level after exiting the range
 pair counting subroutine.

SIGMAX Maximum stress.

SIGMIN Minimum stress.

STMAX Maximum applied stress.

STMIN Minimum applied stress.

SUMDEL Summation of damage for a flight.

SUMENN Accumulated total of applied cycles. (Summation of ENN).

SUMNC Accumulated cycle ratio. (Summation of ENN/CYCLES).

SUMR Summation of $R(N)$, $N=4,7$.

SUMRN Summation of $nR(N)$, $N=4,7$. $n=4$.

SUMR2 Summation of $R(N)^2$, $N=4,7$.

SUMR2N Summation of $nR(N)^2$, $N=4,7$. $n=4$.

SUMR3 Summation of $R(N)^3$, $N=4,7$.

SUMR4 Summation of $R(N)^4$, $N=4,7$.

TITLE1,TITLE2 Identification of the source of the SN data.

TLL Tensile load limit.

TM1,TM2 Material type.

TTYS One-fifth of tensile yield stress.

TYS Tensile yield stress.

T1,T2,T3,T4,T5,T6,T7,T8 Test identifying information.

X Variable equivalent to SIGMIN.

Y Variable equivalent to SIGMAX.

APPENDIX IV

FATIGUE LIFE INPUT DATA FLR SEVERAL MATERIALS

| MATERIAL | YIELD STRESS | STRAIN INTERCEPT | INVERSE OF SLOPE | LIFE, 10 ^I | S-N LIFE COEFFICIENTS | | |
|------------|-----------------|---------------------|---------------------|--------------------------|-----------------------|-------|-------|
| | | | | | A(I) | B(I) | C(I) |
| 2024-T4 | 58. | 0.4 | -1.836 | 4 | -.0020 | .2091 | 62.6 |
| | | | | 5 | -.0032 | .4366 | 51.4 |
| | | | | 6 | -.0035 | .6207 | 42.2 |
| | | | | 7 | -.0042 | .7003 | 36.1 |
| 2219-T851 | 55. | 0.4 | -1.836 | 4 | -.0022 | .2204 | 55.8 |
| | | | | 5 | -.0018 | .3320 | 48.3 |
| | | | | 6 | -.0015 | .4628 | 39.7 |
| | | | | 7 | -.0024 | .6420 | 31.7 |
| 7075-T6 | 72. | 0.4 | -1.836 | 4 | -.0020 | .2801 | 71.7 |
| | | | | 5 | -.0022 | .5154 | 56.3 |
| | | | | 6 | -.0014 | .6141 | 44.6 |
| | | | | 7 | -.0013 | .6838 | 38.1 |
| RQC-100 | 125. | 0.54 | -1.493 | 4 | 0.0 | .2136 | 98.3 |
| | | | | 5 | 0.0 | .2927 | 88.5 |
| | | | | 6 | 0.0 | .3669 | 79.1 |
| | | | | 7 | 0.0 | .4376 | 70.3 |
| Man-Ten | 55. | 1.11 | -1.667 | 4 | 0.0 | .2257 | 63.5 |
| | | | | 5 | 0.0 | .3520 | 53.1 |
| | | | | 6 | 0.0 | .4669 | 43.7 |
| | | | | 7 | 0.0 | .5678 | 35.4 |
| 4340 Steel | 160. | 0.4 | -1.836 | 4 | -.0002 | .2567 | 162.4 |
| | | | | 5 | -.0007 | .5248 | 126.9 |
| | | | | 6 | -.0005 | .5557 | 113.5 |
| | | | | 7 | -.0005 | .5557 | 108.5 |
| Ti-6-4 | 158. | 0.4 | -1.836 | 4 | -.0009 | .2368 | 154.2 |
| | | | | 5 | -.0006 | .4640 | 110.3 |
| | | | | 6 | -.0000 | .4650 | 88.9 |
| | | | | 7 | .0001 | .4752 | 84.2 |